

# A New Sum-of-Squares Design Framework for Robust Control of Polynomial Fuzzy Systems with Uncertainties

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**Abstract**—This paper presents a new sum-of-squares (SOS, for brevity) design framework for robust control of polynomial fuzzy systems with uncertainties. Two kinds of robust stabilization conditions are derived in terms of SOS. One is global SOS robust stabilization conditions that guarantee the global and asymptotical stability of polynomial fuzzy control systems. The other is semi-global SOS robust stabilization conditions. The latter is available for very complicated systems that are difficult to guarantee the global and asymptotical stability of polynomial fuzzy control systems. The main feature of all the SOS robust stabilization conditions derived in this paper are to be expressed as non-convex formulations with respect to polynomial Lyapunov function parameters and polynomial feedback gains. Since a typical transformation from non-convex SOS design conditions to convex SOS design conditions often results in some conservative issues, the new design framework presented in this paper gives key ideas to avoid the conservative issues. The first key idea is that we directly solve non-convex SOS design conditions without applying the typical transformation. The second key idea is that we bring a so-called copositivity concept. These ideas provide some advantages in addition to relaxations. To solve our SOS robust stabilization conditions efficiently, we introduce a gradient algorithm formulated as a minimizing optimization problem of the upper bound of the time derivative of an SOS polynomial that can be regarded as a candidate of polynomial Lyapunov functions. Three design examples are provided to illustrate the validity and applicability of the proposed design framework. The examples demonstrate advantages of our new SOS design framework for the existing LMI approaches and the existing convex SOS approach.

**Index Terms**—copositivity, polynomial Lyapunov function, polynomial fuzzy system with uncertainty, robust stabilization, sum of squares.

## I. INTRODUCTION

**T**ODAY there exists a large body of literature on Takagi-Sugeno (T-S) fuzzy model-based control [1]. Especially, linear matrix inequalities (LMIs) based designs, e.g., [2], [3], have been paid a lot of attention after LMI-based designs have been discussed in [4]-[6]. A key feature of the approach is that it renders simple, natural and effective design procedures as alternatives or supplements to other nonlinear control techniques

(e.g., [7]) that require special and rather involved knowledge. The LMI-based design approaches entail obtaining numerical solutions by convex optimization methods such as the interior point method [8].

Though LMI-based approaches have enjoyed great success and popularity, there still exist a large number of design problems that either cannot be represented in terms of LMIs, or the results obtained through LMIs are sometimes conservative. Recently, as a post-LMI framework, an SOS based approach has received a great deal of attention in control of nonlinear systems using polynomial fuzzy systems and controllers, which includes the well-known Takagi-Sugeno fuzzy systems and controllers as special cases. An SOS approach to polynomial fuzzy control system designs has first presented in [9]-[13]. It can be seen that SOS approaches [9]-[22] provide more extensive and/or relaxed results for the existing LMI approaches [2], [3], [23]-[35] to T-S fuzzy model and control. However, there exists a very few literature on SOS-based robust control designs for polynomial fuzzy systems with uncertainties. To the best of our knowledge, an SOS-based robust control design for polynomial fuzzy systems with uncertainties has been discussed only in [36]. The most important point of SOS-based design conditions is that, to obtain convex SOS design conditions, the existing SOS-based design conditions [9]-[20] utilize a typical transformation from non-convex SOS design conditions to convex SOS design conditions. However, the transformation often results in some conservative issues although no such conservatism exists in LMI transformation cases. In [36], the typical transformation is employed to obtain convex SOS robust stabilization conditions. Furthermore, not only the conservative issues but also other two difficulties are found in the existing SOS approach. One is a restrictive polynomial Lyapunov function setting that leads to conservative stability results. The other is that the stability does not generally hold globally in the existing SOS approach. These will be concretely discussed in Remarks 2 and 3. This paper gives new ideas to solve the conservative issues and the difficulties in the existing SOS approach.

This paper presents a new SOS design framework for robust control of polynomial fuzzy systems with uncertainties. The framework gives key ideas to avoid the conservative issues. The first key idea is that we directly solve non-convex SOS design conditions without applying the typical transformation. The second key idea is that we bring a so-called copositivity concept. These ideas provide some advantages in addition to

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relaxations. To solve our SOS robust stabilization conditions efficiently, we introduce a gradient algorithm formulated as a minimizing optimization problem of the upper bound of the time derivative of an SOS polynomial that can be regarded as a candidate of polynomial Lyapunov functions.

The rest of the paper is organized as follows. Section II recalls a polynomial fuzzy system defined in [9]-[13] and defines a polynomial fuzzy system with uncertainty. Sections III and IV give a new SOS framework for robust control, i.e., robust stabilization conditions to design a robust fuzzy controller and an algorithm to solve them, respectively. Section V entails two design examples to demonstrate the validity and applicability of the proposed design framework. The examples demonstrate advantages of our SOS robust stabilization conditions for the existing LMI approaches and the existing convex SOS approach. Sections VI and VII present semi-global robust stabilization conditions and their design example, respectively. The design example deals with a kind of unmanned aerial vehicles (UAVs) that is a very complicated system with high nonlinearity.

It is assumed that all the matrices and vectors in this paper have appropriate dimensions.  $\mathbf{P} \succ \mathbf{0}$  ( $\mathbf{P} \succeq \mathbf{0}$ ) means that  $\mathbf{P}$  is a positive definite matrix (positive semi-definite matrix).

## II. POLYNOMIAL FUZZY SYSTEM WITH UNCERTAINTIES

Consider the following nonlinear system:

$$\dot{\mathbf{x}}(t) = \mathbf{f}(\mathbf{x}(t), \mathbf{u}(t)), \quad (1)$$

where  $f$  is a smooth nonlinear function such that  $f(\mathbf{0}, \mathbf{0}) = \mathbf{0}$ .  $\mathbf{x}(t) = [x_1(t) \ x_2(t) \ \cdots \ x_n(t)]^T$  is the state vector and  $\mathbf{u}(t) = [u_1(t) \ u_2(t) \ \cdots \ u_m(t)]^T$  is the input vector. Based on the sector nonlinearity concept [2], we can exactly represent (1) with the following T-S fuzzy model [37] (globally or at least semi-globally).

### Model Rule $i$ :

If  $z_1(t)$  is  $M_{i1}$  and  $\cdots$  and  $z_p(t)$  is  $M_{ip}$

$$\text{then } \dot{\mathbf{x}}(t) = \mathbf{A}_i \mathbf{x}(t) + \mathbf{B}_i \mathbf{u}(t) \quad i = 1, 2, \dots, r, \quad (2)$$

where  $z_j(t)$  ( $j = 1, 2, \dots, p$ ) is the premise variable. The membership function,  $M_{ij}$ , denotes the  $j$ th premise variable component in the  $i$ th *Model Rule*.  $r$  denotes the number of *Model Rules*. Each  $z_j(t)$  is a measurable time-varying quantity that may be states, measurable external variables and/or time.

The overall dynamics of the system is represented by fuzzy blending of the linear system models. That is, the defuzzification process of the T-S model (2) can be represented as

$$\begin{aligned} \dot{\mathbf{x}}(t) &= \frac{\sum_{i=1}^r w_i(\mathbf{z}(t)) \{ \mathbf{A}_i \mathbf{x}(t) + \mathbf{B}_i \mathbf{u}(t) \}}{\sum_{i=1}^r w_i(\mathbf{z}(t))} \\ &= \sum_{i=1}^r h_i(\mathbf{z}(t)) \{ \mathbf{A}_i \mathbf{x}(t) + \mathbf{B}_i \mathbf{u}(t) \}, \end{aligned} \quad (3)$$

where

$$\mathbf{z}(t) = [z_1(t) \ \cdots \ z_p(t)]$$

and

$$w_i(\mathbf{z}(t)) = \prod_{j=1}^p M_{ij}(z_j(t)).$$

Since the number of *Model Rule* that fire for all  $t$  is larger than or equal to one in general, the following relations hold.

$$\sum_{i=1}^r w_i(\mathbf{z}(t)) > 0, \quad w_i(\mathbf{z}(t)) \geq 0, \quad i = 1, 2, \dots, r.$$

Hence,

$$h_i(\mathbf{z}(t)) = \frac{w_i(\mathbf{z}(t))}{\sum_{i=1}^r w_i(\mathbf{z}(t))} \geq 0, \quad \sum_{i=1}^r h_i(\mathbf{z}(t)) = 1.$$

In [9] and [12], we proposed a new type of fuzzy model with polynomial model consequence, i.e., fuzzy model whose consequent parts are represented by polynomials. Using the sector nonlinearity concept [2], we exactly represent (1) with the following polynomial fuzzy model (4) even if the nonlinear system (1) contains polynomial elements. The main difference between the T-S fuzzy model [37] and the polynomial fuzzy model is consequent part representation. The fuzzy model (4) has a polynomial model consequence.

### Model Rule $i$ :

If  $z_1(t)$  is  $M_{i1}$  and  $\cdots$  and  $z_p(t)$  is  $M_{ip}$

$$\text{then } \dot{\mathbf{x}}(t) = \mathbf{A}_i(\mathbf{x}(t)) \hat{\mathbf{x}}(\mathbf{x}(t)) + \mathbf{B}_i(\mathbf{x}(t)) \mathbf{u}(t), \quad (4)$$

where  $i = 1, 2, \dots, r$ .  $r$  denotes the number of *Model Rules*.  $\hat{\mathbf{x}}(\mathbf{x}(t))$  is a column vector whose entries are all monomials in  $\mathbf{x}(t)$ . That is,  $\hat{\mathbf{x}}(\mathbf{x}(t)) \in \mathbf{R}^N$  is an  $N \times 1$  vector of monomials in  $\mathbf{x}(t)$ . A monomial in  $\mathbf{x}(t)$  is a function of the form  $x_1^{\alpha_1} x_2^{\alpha_2} \cdots x_n^{\alpha_n}$ , where  $\alpha_1, \alpha_2, \dots, \alpha_n$  are nonnegative integers.  $\mathbf{A}_i(\mathbf{x}(t)) \in \mathbf{R}^{n \times N}$  and  $\mathbf{B}_i(\mathbf{x}(t)) \in \mathbf{R}^{n \times m}$  are polynomial matrices in  $\mathbf{x}(t)$ . Therefore,  $\mathbf{A}_i(\mathbf{x}(t)) \hat{\mathbf{x}}(\mathbf{x}(t)) + \mathbf{B}_i(\mathbf{x}(t)) \mathbf{u}(t)$  is a polynomial vector. Thus, the polynomial fuzzy model (4) has a polynomial in each consequent part. We assume that

$$\hat{\mathbf{x}}(\mathbf{x}(t)) = \mathbf{0} \text{ iff } \mathbf{x}(t) = \mathbf{0}$$

throughout this paper.

The defuzzification process of the model (4) can be represented as

$$\dot{\mathbf{x}}(t) = \sum_{i=1}^r h_i(\mathbf{z}(t)) \{ \mathbf{A}_i(\mathbf{x}(t)) \hat{\mathbf{x}}(\mathbf{x}(t)) + \mathbf{B}_i(\mathbf{x}(t)) \mathbf{u}(t) \}. \quad (5)$$

The polynomial fuzzy model is an extension of the T-S fuzzy model. The extension bring us some advantages [12]. One is that SOS stabilization conditions provides more relaxed results than the existing LMI stabilization conditions. Another advance is that original nonlinear systems with polynomial terms can be exactly and globally represented by polynomial fuzzy models although the T-S fuzzy models are sometimes not global models for the original nonlinear systems with polynomial terms.

**Remark 1.** Stability conditions for the T-S fuzzy system have been mainly represented in terms of LMIs [2]. Hence, the LMI stability conditions can be solved numerically and efficiently by interior point algorithms, e.g., by LMI solvers. On the other hand, the convex SOS conditions in [9]-[20], [36] for polynomial fuzzy systems are represented as convex SOS problems. Clearly, the problems can not be directly solved by LMI solvers, but they can be solved via an SOS solver (SOSOFT [38], SOSTOOLS [39], etc.) and an SDP solver [40], [41].

This paper focuses on stabilization of the polynomial fuzzy model with uncertainties. Hence, we define a polynomial fuzzy model with uncertainties as follows.

**Model Rule  $i$ :**

If  $z_1(t)$  is  $M_{i1}$  and  $\dots$  and  $z_p(t)$  is  $M_{ip}$  then

$$\begin{aligned} \dot{\mathbf{x}}(t) = & \{ \mathbf{A}_i(\mathbf{x}(t)) \\ & + \mathbf{D}_{ai}(\mathbf{x}(t))\Delta_{ai}(\mathbf{x}(t))\mathbf{E}_{ai}(\mathbf{x}(t)) \} \hat{\mathbf{x}}(\mathbf{x}(t)) \\ & + \{ \mathbf{B}_i(\mathbf{x}(t)) \\ & + \mathbf{D}_{bi}(\mathbf{x}(t))\Delta_{bi}(\mathbf{x}(t))\mathbf{E}_{bi}(\mathbf{x}(t)) \} \mathbf{u}(t), \end{aligned} \quad (6)$$

where  $i = 1, 2, \dots, r$ .  $\mathbf{D}_{ai}(\mathbf{x}(t))$ ,  $\mathbf{D}_{bi}(\mathbf{x}(t))$ ,  $\mathbf{E}_{ai}(\mathbf{x}(t))$  and  $\mathbf{E}_{bi}(\mathbf{x}(t))$  are polynomial matrices in  $\mathbf{x}(\mathbf{x}(t))$ .  $\Delta_{ai}(\mathbf{x}(t))$  and  $\Delta_{bi}(\mathbf{x}(t))$  denote uncertain matrices in  $\mathbf{x}(t)$  and satisfy

$$\| \Delta_{ai}(\mathbf{x}(t)) \| \leq \beta_{ai}(\mathbf{x}(t)), \quad (7)$$

$$\| \Delta_{bi}(\mathbf{x}(t)) \| \leq \beta_{bi}(\mathbf{x}(t)), \quad (8)$$

where  $\beta_{ai}(\mathbf{x}(t))$  and  $\beta_{bi}(\mathbf{x}(t))$  denote the upper bound of the norm of the uncertainties.

The defuzzification process of the model (6) can be represented as

$$\begin{aligned} \dot{\mathbf{x}}(t) = & \sum_{i=1}^r h_i(z(t)) \{ \mathbf{A}_i(\mathbf{x}(t)) \hat{\mathbf{x}}(\mathbf{x}(t)) + \mathbf{B}_i(\mathbf{x}(t)) \mathbf{u}(t) \\ & + \mathbf{D}_{ai}(\mathbf{x}(t)) \Delta_{ai}(\mathbf{x}(t)) \mathbf{E}_{ai}(\mathbf{x}(t)) \hat{\mathbf{x}}(\mathbf{x}(t)) \\ & + \mathbf{D}_{bi}(\mathbf{x}(t)) \Delta_{bi}(\mathbf{x}(t)) \mathbf{E}_{bi}(\mathbf{x}(t)) \mathbf{u}(t) \}. \end{aligned} \quad (9)$$

From now, to lighten the notation, we will drop the notation with respect to time  $t$ . For instance, we will employ  $\mathbf{x}$  and  $\hat{\mathbf{x}}(\mathbf{x})$  instead of  $\mathbf{x}(t)$  and  $\hat{\mathbf{x}}(\mathbf{x}(t))$ , respectively. Thus, we drop the notation with respect to time  $t$ , but it should be kept in mind that  $\mathbf{x}$  and  $\hat{\mathbf{x}}(\mathbf{x})$  means  $\mathbf{x}(t)$  and  $\hat{\mathbf{x}}(\mathbf{x}(t))$ , respectively.

For the model (9), we design the following fuzzy controller.

$$\mathbf{u} = - \sum_{i=1}^r h_i(z) \mathbf{F}_i(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \quad (10)$$

A convex SOS robust design condition for the control system consisting of (9) and (10) was presented in [36]. However, as will be mentioned in Remarks 2 and 3, some disadvantages exist in the existing SOS approaches [9]-[20] [36].

**Remark 2.** In [9]-[20] and [36], the Lyapunov function candidate (11) is used.

$$V(\mathbf{x}) = \hat{\mathbf{x}}^T(\mathbf{x}) \mathbf{X}^{-1}(\tilde{\mathbf{x}}) \hat{\mathbf{x}}(\mathbf{x}), \quad (11)$$

where  $\mathbf{X}(\tilde{\mathbf{x}})$  is a polynomial matrix in  $\tilde{\mathbf{x}}$ . If  $\hat{\mathbf{x}}(\mathbf{x}) = \mathbf{x}$  and  $\mathbf{X}^{-1}(\tilde{\mathbf{x}})$  is a constant matrix, then (11) reduces to the quadratic Lyapunov function. The zero equilibrium is asymptotically stable when the Lyapunov function exists. However, the globality is not guaranteed. The stability holds globally only if  $\mathbf{X}^{-1}(\tilde{\mathbf{x}})$  is a constant matrix. The important point is that, to avoid introducing non-convex condition,  $\tilde{\mathbf{x}}$  in the polynomial matrix  $\mathbf{X}(\tilde{\mathbf{x}})$  is defined as follows. Let  $\mathbf{K} = \{k_1, k_2, \dots, k_m\}$  denote the row indices of  $\mathbf{B}_i(\mathbf{x})$  whose corresponding row is equal to zero, and define  $\tilde{\mathbf{x}} = (x_{k_1}, x_{k_2}, \dots, x_{k_m})$  using the  $\mathbf{K}$ . In other words, to avoid introducing non-convex condition, it is assumed in the literature that  $\mathbf{X}(\tilde{\mathbf{x}})$  only depends on states  $\tilde{\mathbf{x}}$  whose dynamics is not directly affected by the control input, namely states whose corresponding rows in  $\mathbf{B}_i(\mathbf{x})$  are zero. The restriction caused by  $\tilde{\mathbf{x}}$  depends on the  $\mathbf{B}_i(\mathbf{x})$  matrices and it leads to some conservative stability results. A new SOS framework that will be presented in Section III permits a non-restrictive polynomial Lyapunov function setting.

**Remark 3.** As mentioned in Remark 2, (11) is employed as a candidate Lyapunov function. The transformation from non-convex conditions to convex conditions is carried out as follows. The time derivative of  $V(\mathbf{x})$  along the feedback system trajectory, that consists of (5) and (10), can be represented by the general form.

$$\dot{V}(\mathbf{x}) = \hat{\mathbf{x}}^T(\mathbf{x}) \mathbf{S}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) < 0, \quad (12)$$

where  $\mathbf{S}(\mathbf{x})$  is a non-convex polynomial matrix since it has cross terms with respect to  $\mathbf{X}^{-1}(\tilde{\mathbf{x}})$  and  $\mathbf{F}_i(\mathbf{x})$ . The transformation is carried out by dropping  $\hat{\mathbf{x}}(\mathbf{x})$  off from both side of the inequality and by multiplying the dropped inequality on the left and right by  $\mathbf{X}(\tilde{\mathbf{x}})$ . As a result of the transformation, we have the following convex condition with respect to  $\mathbf{X}(\tilde{\mathbf{x}})$  and  $\mathbf{M}_i(\mathbf{x})$ , where  $\mathbf{M}_i(\mathbf{x}) = \mathbf{F}_i(\mathbf{x}) \mathbf{X}(\tilde{\mathbf{x}})$ .

$$-\mathbf{X}(\tilde{\mathbf{x}}) \mathbf{S}(\mathbf{x}) \mathbf{X}(\tilde{\mathbf{x}}) \succ \mathbf{0}.$$

Finally, we arrive at the convex SOS condition,

$$-\mathbf{v}^T \{ \mathbf{X}(\tilde{\mathbf{x}}) \mathbf{S}(\mathbf{x}) \mathbf{X}(\tilde{\mathbf{x}}) + \epsilon(\mathbf{x}) \mathbf{I} \} \mathbf{v} \text{ is SOS,}$$

where  $\epsilon(\mathbf{x})$  is a slack variable (a radially unbounded positive definite polynomial) to keep the positivity of the SOS condition. In the transformation, we utilize the fact that  $-\mathbf{S}(\mathbf{x}) \succ \mathbf{0} \Rightarrow -\hat{\mathbf{x}}^T(\mathbf{x}) \mathbf{S}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) > 0$ . However, it should be emphasized that this is a sufficient condition, i.e., in general, it is not always satisfied that  $-\hat{\mathbf{x}}^T(\mathbf{x}) \mathbf{S}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) > 0 \Rightarrow -\mathbf{S}(\mathbf{x}) \succ \mathbf{0}$ . It becomes a necessary and sufficient condition only if  $\mathbf{S}(\mathbf{x})$  is a constant matrix and  $\hat{\mathbf{x}}(\mathbf{x}) = \mathbf{x}$ . Only in the case, no conservatism exists. In the LMI case [2], this path is always equivalent since  $\mathbf{S}(\mathbf{x})$  is a constant matrix and  $\hat{\mathbf{x}}(\mathbf{x}) = \mathbf{x}$ . Thus, this conservative path in the convex SOS transformation often causes conservative results although this path is always equivalent in the LMI case. In [36], the same transformation is employed to obtain convex SOS robust stabilization conditions. A new SOS framework that will be presented in Section III can avoid this main problem.

A new SOS framework that will be presented in Section III can completely avoid the two problems mentioned in Remarks

2 and 3. The utility of the new SOS design framework will be demonstrated in design examples.

### III. SOS STABILIZATION CONDITIONS

Section III presents SOS stabilization conditions based on copositivity concept.

If (13) holds, the matrix  $\mathbf{J} = [J_{ij}] \in \mathbf{R}^{\ell \times \ell}$  is copositive.

$$\mathbf{y}^T \mathbf{J} \mathbf{y} = \sum_{i=1}^{\ell} \sum_{j=1}^{\ell} y_i y_j J_{ij} \geq 0, \quad (13)$$

where  $\mathbf{y} = [y_1, y_2, \dots, y_{\ell}]^T \in \mathbf{R}^{\ell}$  and  $y_i \geq 0$ . Since checking copositivity of a matrix is a co-NP complete problem, we take a technique for copositivity checking relaxation [39].

#### Corollary 1. [39]

A relaxation is to introduce  $y_i = \hat{y}_i^2$  and to check whether (14) is satisfied or not.

$$\mathbf{Q}^s(\hat{\mathbf{y}}) = \left( \sum_{k=1}^{\ell} \hat{y}_k^2 \right)^s \sum_{i=1}^{\ell} \sum_{j=1}^{\ell} \hat{y}_i^2 \hat{y}_j^2 J_{ij} \text{ is SOS}, \quad (14)$$

where  $\hat{\mathbf{y}} = [\hat{y}_1, \hat{y}_2, \dots, \hat{y}_{\ell}]^T$  and  $s$  is a nonnegative integer.

By using the copositivity checking relaxation, we derive SOS robust stabilization conditions that are different from the SOS robust stabilization conditions in [36]. Theorem 1 presents the SOS robust stabilization conditions.

**Theorem 1.** *If there exist a polynomial  $V(\mathbf{x})$ , polynomial matrices  $\mathbf{F}_j(\mathbf{x})$  and polynomials  $\bar{g}_{ij}(\mathbf{x})$  such that (15) ~ (17) are satisfied with  $\alpha < 0$  and  $\lambda > 0$ , the polynomial fuzzy controller (10) stabilizes the system (9), and  $V(\mathbf{x})$  becomes a Lyapunov function.*

$$V(\mathbf{x}) - \epsilon(\mathbf{x}) \text{ is SOS}, \quad (15)$$

$$\left( \sum_{k=1}^r \hat{h}_k^2 \right)^s \sum_{i=1}^r \sum_{j=1}^r \hat{h}_i^2 \hat{h}_j^2 \{ -\bar{\Lambda}_{ij}(\mathbf{x}) + \alpha V(\mathbf{x}) \} \text{ is SOS}, \quad (16)$$

$$\mathbf{v}_1^T \mathbf{L}_{ij}(\lambda, \mathbf{x}) \mathbf{v}_1 \text{ is SOS}, \quad (17)$$

where  $\mathbf{v}_1$  denotes vector that is independent of  $\mathbf{x}$ .  $\epsilon(\mathbf{x})$  is a radially unbounded positive definite polynomial and  $s$  is a non-negative integer.

$$\bar{\Lambda}_{ij}(\mathbf{x}) = \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \{ \mathbf{A}_i(\mathbf{x}) - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \} \hat{\mathbf{x}}(\mathbf{x}) + \bar{g}_{ij}(\mathbf{x}), \quad (18)$$

$$\mathbf{L}_{ij}(\lambda, \mathbf{x}) = \begin{bmatrix} \lambda \bar{g}_{ij}(\mathbf{x}) & * & * & * & * \\ \lambda \mathbf{D}_{ai}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T & 2\mathbf{I} & \mathbf{0} & \mathbf{0} & \mathbf{0} \\ \lambda \mathbf{D}_{bi}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T & \mathbf{0} & 2\mathbf{I} & \mathbf{0} & \mathbf{0} \\ \beta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) & \mathbf{0} & \mathbf{0} & 2\mathbf{I} & \mathbf{0} \\ \beta_{bi}(\mathbf{x}) \mathbf{E}_{bi}(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) & \mathbf{0} & \mathbf{0} & \mathbf{0} & 2\mathbf{I} \end{bmatrix}. \quad (19)$$

The asterisk \* denotes the transposed elements (matrices) for symmetric positions.

*Proof:*

Consider a candidate of Lyapunov functions  $V(\mathbf{x})$ . The time derivative of  $V(\mathbf{x})$  is given as

$$\dot{V}(\mathbf{x}) = \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \dot{\mathbf{x}}. \quad (20)$$

By substituting the closed loop dynamics consisting of (9) and (10) into (20), the time derivative of  $V(\mathbf{x})$  along the trajectory becomes

$$\begin{aligned} \dot{V}(\mathbf{x}) &= \sum_{i=1}^r \sum_{j=1}^r h_i h_j \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \{ \mathbf{A}_i(\mathbf{x}) - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \\ &\quad + \mathbf{D}_{ai}(\mathbf{x}) \Delta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \\ &\quad - \mathbf{D}_{bi}(\mathbf{x}) \Delta_{bi}(\mathbf{x}) \mathbf{E}_{bi}(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \} \hat{\mathbf{x}}(\mathbf{x}) \\ &= \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left\{ \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) \right. \\ &\quad \left. - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) + \zeta_i(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}) \right\}, \end{aligned}$$

where

$$\zeta_i(\mathbf{x}) = \begin{bmatrix} \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \mathbf{D}_{ai}(\mathbf{x}) & \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \mathbf{D}_{bi}(\mathbf{x}) \end{bmatrix},$$

$$\boldsymbol{\eta}_{ij}(\mathbf{x}) = \begin{bmatrix} \Delta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \\ -\Delta_{bi}(\mathbf{x}) \mathbf{E}_{bi}(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \end{bmatrix}.$$

Note that

$$\begin{aligned} \lambda \zeta_i(\mathbf{x}) \zeta_i^T(\mathbf{x}) &+ \frac{1}{\lambda} \boldsymbol{\eta}_{ij}^T(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}) \\ &\geq \zeta_i(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}) + \boldsymbol{\eta}_{ij}^T(\mathbf{x}) \zeta_i^T(\mathbf{x}) \end{aligned}$$

for any  $\lambda > 0$ . In addition, since  $\zeta_i(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}) = \boldsymbol{\eta}_{ij}^T(\mathbf{x}) \zeta_i^T(\mathbf{x})$ , we have the following relation.

$$\lambda \zeta_i(\mathbf{x}) \zeta_i^T(\mathbf{x}) + \frac{1}{\lambda} \boldsymbol{\eta}_{ij}^T(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}) \geq 2 \zeta_i(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}).$$

Hence

$$\begin{aligned} &\zeta_i(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}) \\ &\leq \frac{\lambda}{2} \zeta_i(\mathbf{x}) \zeta_i^T(\mathbf{x}) + \frac{1}{2\lambda} \boldsymbol{\eta}_{ij}^T(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}) \\ &= \frac{\lambda}{2} \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \mathbf{D}_{ai}(\mathbf{x}) \mathbf{D}_{ai}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T \\ &\quad + \frac{\lambda}{2} \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \mathbf{D}_{bi}(\mathbf{x}) \mathbf{D}_{bi}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T \\ &\quad + \frac{1}{2\lambda} \hat{\mathbf{x}}^T(\mathbf{x}) \mathbf{E}_{ai}^T(\mathbf{x}) \Delta_{ai}^T(\mathbf{x}) \Delta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \\ &\quad + \frac{1}{2\lambda} \hat{\mathbf{x}}^T(\mathbf{x}) \mathbf{F}_j^T(\mathbf{x}) \mathbf{E}_{bi}^T(\mathbf{x}) \Delta_{bi}^T(\mathbf{x}) \\ &\quad \quad \times \Delta_{bi}(\mathbf{x}) \mathbf{E}_{bi}(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \\ &\leq \boldsymbol{\Pi}_{ij}(\lambda, \mathbf{x}), \end{aligned}$$

where

$$\begin{aligned} \boldsymbol{\Pi}_{ij}(\lambda, \mathbf{x}) &= \frac{\lambda}{2} \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \mathbf{D}_{ai}(\mathbf{x}) \mathbf{D}_{ai}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T \\ &\quad + \frac{\lambda}{2} \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \mathbf{D}_{bi}(\mathbf{x}) \mathbf{D}_{bi}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T \\ &\quad + \frac{1}{2\lambda} \beta_{ai}^2(\mathbf{x}) \hat{\mathbf{x}}^T(\mathbf{x}) \mathbf{E}_{ai}^T(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \\ &\quad + \frac{1}{2\lambda} \beta_{bi}^2(\mathbf{x}) \hat{\mathbf{x}}^T(\mathbf{x}) \mathbf{F}_j^T(\mathbf{x}) \mathbf{E}_{bi}^T(\mathbf{x}) \mathbf{E}_{bi}(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}). \end{aligned}$$

From the above inequality, we have

$$\begin{aligned} \dot{V}(\mathbf{x}) &= \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left\{ \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) \right. \\ &\quad \left. - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) + \zeta_i(\mathbf{x}) \boldsymbol{\eta}_{ij}(\mathbf{x}) \right\} \\ &\leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left\{ \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) \right. \\ &\quad \left. - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) + \boldsymbol{\Pi}_{ij}(\lambda, \mathbf{x}) \right\}. \end{aligned}$$

We introduce a polynomial  $\bar{g}_{ij}(\mathbf{x})$  satisfying

$$\sum_{i=1}^r \sum_{j=1}^r h_i h_j \boldsymbol{\Pi}_{ij}(\lambda, \mathbf{x}) \leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j \bar{g}_{ij}(\mathbf{x}). \quad (21)$$

Then, we have

$$\begin{aligned} \dot{V}(\mathbf{x}) &\leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left\{ \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) \right. \\ &\quad \left. - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) + \bar{g}_{ij}(\mathbf{x}) \right\}. \end{aligned} \quad (22)$$

To show that  $\dot{V}(\mathbf{x}) < 0$  at  $\mathbf{x} \neq 0$ , we consider the condition satisfying  $\dot{V}(\mathbf{x}) \leq \alpha V(\mathbf{x})$ , where  $\alpha < 0$ . That is,

$$\begin{aligned} \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left\{ \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) \right. \\ \left. + \bar{g}_{ij}(\mathbf{x}) \right\} - \alpha V(\mathbf{x}) \leq 0. \end{aligned}$$

By applying the copositivity presented in Lemma 1, we obtain (16).

On the other hand, from the inequality (21) and  $\lambda > 0$ , we obtain

$$\sum_{i=1}^r \sum_{j=1}^r h_i h_j \{ \lambda \bar{g}_{ij}(\mathbf{x}) - \lambda \boldsymbol{\Pi}_{ij}(\lambda, \mathbf{x}) \} \geq 0. \quad (23)$$

Using schur complement, (23) can be converted to

$$\sum_{i=1}^r \sum_{j=1}^r h_i h_j \mathbf{L}_{ij}(\lambda, \mathbf{x}) \geq \mathbf{0}, \quad (24)$$

where

$$\mathbf{L}_{ij}(\lambda, \mathbf{x}) = \begin{bmatrix} \lambda \bar{g}_{ij}(\mathbf{x}) & * & * & * & * \\ \lambda \mathbf{D}_{ai}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T & 2\mathbf{I} & \mathbf{0} & \mathbf{0} & \mathbf{0} \\ \lambda \mathbf{D}_{bi}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T & \mathbf{0} & 2\mathbf{I} & \mathbf{0} & \mathbf{0} \\ \beta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) & \mathbf{0} & \mathbf{0} & 2\mathbf{I} & \mathbf{0} \\ \beta_{bi}(\mathbf{x}) \mathbf{E}_{bi}(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) & \mathbf{0} & \mathbf{0} & \mathbf{0} & 2\mathbf{I} \end{bmatrix}.$$

The condition (24) holds if (17) is satisfied.  $\blacksquare$

**Theorem 2.** Assume that  $\Delta_{bi}(\mathbf{x}) = \mathbf{0}$  for all  $i$ , i.e., there are no uncertainties with respect to the input terms. Then, the SOS robust stabilization conditions in Theorem 1 become simple. If there exist a polynomial function  $V(\mathbf{x})$ , polynomial matrices  $\mathbf{F}_j(\mathbf{x})$  and polynomials  $\bar{g}_i(\mathbf{x})$  such that (25) ~ (27) are satisfied with  $\alpha < 0$  and  $\lambda > 0$ , the polynomial fuzzy controller (10) stabilizes the system (9).

$$V(\mathbf{x}) - \epsilon(\mathbf{x}) \text{ is SOS}, \quad (25)$$

$$\left( \sum_{k=1}^r \hat{h}_k^2 \right)^s \sum_{i=1}^r \sum_{j=1}^r \hat{h}_i^2 \hat{h}_j^2 \{ -\bar{\Lambda}_{ij}(\mathbf{x}) + \alpha V(\mathbf{x}) \} \text{ is SOS}, \quad (26)$$

$$\mathbf{v}_1^T \begin{bmatrix} \lambda \bar{g}_i(\mathbf{x}) & * & * \\ \lambda \mathbf{D}_{ai}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T & 2\mathbf{I} & \mathbf{0} \\ \beta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) & \mathbf{0} & 2\mathbf{I} \end{bmatrix} \mathbf{v}_1 \text{ is SOS}, \quad (27)$$

where  $\epsilon(\mathbf{x})$  is a radially unbounded positive definite polynomial,  $s$  is a non-negative integer, and

$$\begin{aligned} \bar{\Lambda}_{ij}(\mathbf{x}) &= \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \{ \mathbf{A}_i(\mathbf{x}) - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x}) \} \hat{\mathbf{x}}(\mathbf{x}) \\ &\quad + \bar{g}_i(\mathbf{x}). \end{aligned} \quad (28)$$

*Proof:* The proof is omitted since it is directly obtained from Theorem 1. In this case, (17) is reduced to (27).  $\blacksquare$

#### IV. ALGORITHM TO SOLVE SOS CONDITIONS

Section IV presents an algorithm to solve the SOS robust stabilization conditions given in Section III. The algorithm is based on a gradient algorithm formulated as a minimizing optimization problem of the upper bound of the time derivative of an SOS polynomial that can be regarded as a candidate of polynomial Lyapunov functions.

We first explain the outline of its key idea below.

##### A. Key Idea

Consider the non-convex condition

$$\phi_g(\mathbf{x}) \phi_h(\mathbf{x}) \prec 0, \quad (29)$$

where  $\phi_g(\mathbf{x})$  and  $\phi_h(\mathbf{x})$  are polynomial matrices in  $\mathbf{x}$  and both of them are decision variables (matrices). The problem is to find a solution satisfying (29). With a positive definite polynomial matrix  $\psi(\mathbf{x})$  in  $\mathbf{x}$ , the problem (29) may be converted as

$$-\phi_g(\mathbf{x}) \phi_h(\mathbf{x}) + \alpha \psi(\mathbf{x}) \geq 0. \quad (30)$$

If we get a solution of (30) with  $\alpha < 0$ , the problem (29) is feasible. Regularly, (30) can be converted as

$$-\mathbf{v}^T \{ \phi_g(\mathbf{x}) \phi_h(\mathbf{x}) - \alpha \psi(\mathbf{x}) \} \mathbf{v} \text{ is SOS'}$$

where  $\mathbf{v}$  denotes a vector that is independent of  $\mathbf{x}$ . Note that the SOS condition is bilinear (not convex) with respect to decision variables since there exists the term  $\phi_g(\mathbf{x}) \phi_h(\mathbf{x})$ . Now consider very small perturbations  $\delta \phi_g(\mathbf{x})$ ,  $\delta \phi_h(\mathbf{x})$  and  $\delta \psi(\mathbf{x})$  as in [42], [43]. Since  $\delta \phi_g(\mathbf{x})$  and  $\delta \phi_h(\mathbf{x})$  are very small perturbations, it can be noted with a reasonable approximation that

$$\begin{aligned} \phi_g(\mathbf{x}) \phi_h(\mathbf{x}) &\simeq (\phi_g(\mathbf{x}) + \delta \phi_g(\mathbf{x})) (\phi_h(\mathbf{x}) + \delta \phi_h(\mathbf{x})) \\ &= \phi_g(\mathbf{x}) \phi_h(\mathbf{x}) + \delta \phi_g(\mathbf{x}) \phi_h(\mathbf{x}) \\ &\quad + \phi_g(\mathbf{x}) \delta \phi_h(\mathbf{x}) + \delta \phi_g(\mathbf{x}) \phi_h(\mathbf{x}). \end{aligned}$$

Note that the term,  $\delta \phi_g(\mathbf{x}) \delta \phi_h(\mathbf{x})$ , is an extremely small in comparison with other terms since it is the product term of these small perturbations. Then,  $(\phi_g(\mathbf{x}) + \delta \phi_g(\mathbf{x})) (\phi_h(\mathbf{x}) + \delta \phi_h(\mathbf{x}))$  can be represented as  $\phi_g(\mathbf{x}) \phi_h(\mathbf{x}) + \delta \phi_g(\mathbf{x}) \phi_h(\mathbf{x}) + \phi_g(\mathbf{x}) \delta \phi_h(\mathbf{x})$ . From this fact, we transform

$-\phi_g(\mathbf{x})\phi_h(\mathbf{x}) + \alpha\psi(\mathbf{x}) \succeq 0'$   
 to

$$\begin{aligned}
 & -\mathbf{v}^T \{ \phi_g(\mathbf{x})\phi_h(\mathbf{x}) + \delta\phi_g(\mathbf{x})\phi_h(\mathbf{x}) + \phi_g(\mathbf{x})\delta\phi_h(\mathbf{x}) \\
 & \quad - \alpha\psi(\mathbf{x}) - \alpha\delta\psi(\mathbf{x}) \} \mathbf{v} \text{ is SOS}'. \quad (31)
 \end{aligned}$$

Now we can formulate (31) as a minimizing optimization problem based on convex SOS with respect to  $\delta\phi_g(\mathbf{x})$ ,  $\delta\phi_h(\mathbf{x})$  and  $\delta\psi(\mathbf{x})$ .

$$\begin{aligned}
 & \min_{\delta\phi_g(\mathbf{x}), \delta\phi_h(\mathbf{x}), \delta\psi(\mathbf{x})} \alpha \\
 & \text{subject to}
 \end{aligned}$$

$$\mathbf{v}_1^T \{ \psi(\mathbf{x}) + \delta\psi(\mathbf{x}) - \epsilon(\mathbf{x}) \} \mathbf{v}_1 \text{ is SOS}, \quad (32)$$

$$\begin{aligned}
 & -\mathbf{v}_2^T \{ \phi_g(\mathbf{x})\phi_h(\mathbf{x}) + \delta\phi_g(\mathbf{x})\phi_h(\mathbf{x}) + \phi_g(\mathbf{x})\delta\phi_h(\mathbf{x}) \\
 & \quad - \alpha\psi(\mathbf{x}) - \alpha\delta\psi(\mathbf{x}) \} \mathbf{v}_2 \text{ is SOS}, \quad (33)
 \end{aligned}$$

$$\mathbf{v}_3^T \begin{bmatrix} \epsilon_G \phi_g^T(\mathbf{x})\phi_g(\mathbf{x}) & \delta\phi_g(\mathbf{x}) \\ \delta\phi_g(\mathbf{x}) & \mathbf{I} \end{bmatrix} \mathbf{v}_3 \text{ is SOS}, \quad (34)$$

$$\mathbf{v}_4^T \begin{bmatrix} \epsilon_H \phi_h^T(\mathbf{x})\phi_h(\mathbf{x}) & \delta\phi_h(\mathbf{x}) \\ \delta\phi_h(\mathbf{x}) & \mathbf{I} \end{bmatrix} \mathbf{v}_4 \text{ is SOS}, \quad (35)$$

$$\mathbf{v}_5^T \begin{bmatrix} \epsilon_\psi \psi^T(\mathbf{x})\psi(\mathbf{x}) & \delta\psi(\mathbf{x}) \\ \delta\psi(\mathbf{x}) & \mathbf{I} \end{bmatrix} \mathbf{v}_5 \text{ is SOS}, \quad (36)$$

where  $\mathbf{v}_1 - \mathbf{v}_5$  denote vectors that are independent of  $\mathbf{x}$ .  $\epsilon_G$ ,  $\epsilon_H$  and  $\epsilon_\psi$  are very small positive values.  $\epsilon(\mathbf{x})$  is a radially unbounded positive definite polynomial. (34), (35) and (36) guarantee to keep the assumption that  $\delta\phi_g(\mathbf{x})$ ,  $\delta\phi_h(\mathbf{x})$  and  $\delta\psi(\mathbf{x})$  are very small perturbations, respectively.

Note that the decision variables are  $\delta\phi_g(\mathbf{x})$ ,  $\delta\phi_h(\mathbf{x})$  and  $\delta\psi(\mathbf{x})$  in the minimizing optimization. The minimizing optimization is iteratively performed by substituting the solutions  $\delta\phi_g(\mathbf{x})$ ,  $\delta\phi_h(\mathbf{x})$  and  $\delta\psi(\mathbf{x})$

obtained at the  $N$ th iteration into the iteration law

$$\begin{aligned}
 \phi_g^{N+1}(\mathbf{x}) &= \phi_g^N(\mathbf{x}) + \delta\phi_g(\mathbf{x}), \\
 \phi_h^{N+1}(\mathbf{x}) &= \phi_h^N(\mathbf{x}) + \delta\phi_h(\mathbf{x}), \\
 \psi^{N+1}(\mathbf{x}) &= \psi^N(\mathbf{x}) + \delta\psi(\mathbf{x}).
 \end{aligned}$$

Thus, the decision variables are updated so as to minimize the minimizing parameter  $\alpha$ . As a result,  $\phi_g(\mathbf{x})$ ,  $\phi_h(\mathbf{x})$  and  $\psi(\mathbf{x})$  are iteratively updated from the initial setting ( $\phi_g^0(\mathbf{x})$ ,  $\phi_h^0(\mathbf{x})$  and  $\psi^0(\mathbf{x})$ ) so as to minimize the minimizing parameter  $\alpha$ . The initial setting of  $\phi_g^0(\mathbf{x})$ ,  $\phi_h^0(\mathbf{x})$  and  $\psi^0(\mathbf{x})$  should be sometimes carefully selected. So the grid search will be employed to select the initial setting. If the minimizing optimization problem is feasible with  $\alpha < 0$ , it is a solution of (29), i.e.,  $\phi_g(\mathbf{x})\phi_h(\mathbf{x}) < 0$ .

### B. Algorithm

We can consider

$$V(\mathbf{x}) = \bar{\mathbf{x}}^T(\mathbf{x})\mathbf{P}\bar{\mathbf{x}}(\mathbf{x}),$$

where  $\mathbf{P} \in \mathbf{R}^{\rho \times \rho}$  is a positive definite matrix and  $\bar{\mathbf{x}}(\mathbf{x}) \in \mathbf{R}^\rho$  is a column vector whose entries are all monomials in  $\mathbf{x}$  such that  $\bar{\mathbf{x}}(\mathbf{x}) = \mathbf{0}$  iff  $\mathbf{x} = \mathbf{0}$  and  $\|\bar{\mathbf{x}}(\mathbf{x})\| \rightarrow \infty$  for  $\|\mathbf{x}(\mathbf{x})\| \rightarrow \infty$ . For example, if we choose the vector  $\bar{\mathbf{x}}(\mathbf{x}) = [x_1 \ x_2]$  in the case of  $\mathbf{x} = [x_1 \ x_2]$ ,  $V(\mathbf{x})$  becomes a quadratic Lyapunov

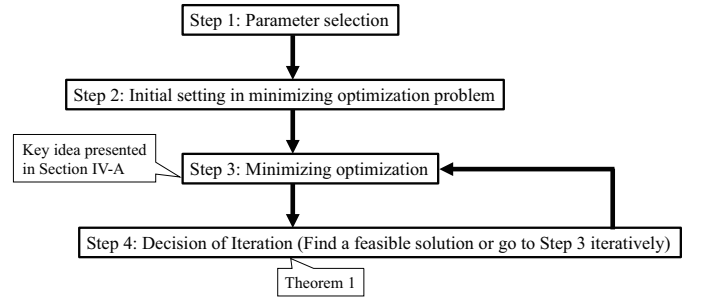


Fig. 1. Outline of algorithm.

function. If  $\bar{\mathbf{x}}(\mathbf{x}) = [x_1^2 \ x_1x_2 \ x_2^2]$  is chosen,  $V(\mathbf{x})$  becomes a 4th-order polynomial Lyapunov function.

The algorithm to solve the SOS conditions consists of four steps. Fig. 1 shows the outline of the algorithm. The key idea mentioned in Section IV-A will be used in Step 3. We check whether the SOS conditions given in Theorem 1 are strictly and exactly feasible or not in Step 4. This algorithm can be regarded as a gradient algorithm formulated as a minimizing optimization problem of the upper bound of the time derivative of the polynomial  $V(\mathbf{x})$ . Table I summarizes main variables and parameters in the minimizing optimization algorithm, where  $p_i$  is the  $i$ -th diagonal element of the positive definite matrix  $\mathbf{P}$ . For simplicity, all the non-diagonal elements of the positive definite matrix  $\mathbf{P}$  are set to zero in the initial setting. However, note that, after performing the algorithm, the nondiagonal elements of the matrix  $\mathbf{P}$  can become non-zero. In fact, the complicated example in Section VII obtains the matrix  $\mathbf{P}$  whose nondiagonal elements are non-zero although the nondiagonal elements of the matrix  $\mathbf{P}$  are set to zero in the initial setting.

TABLE I  
 LIST OF MAIN VARIABLES AND PARAMETERS IN MINIMIZING OPTIMIZATION ALGORITHM.

$N$	number of iteration
$\lambda^{min}, \lambda^{max}$	lower and upper bounds of $\lambda$ satisfying $0 < \lambda^{min} \leq \lambda \leq \lambda^{max}$
$p_i^{min}, p_i^{max}$	lower and upper bounds of $p_i$ satisfying $0 < p_i^{min} \leq p_i \leq p_i^{max}$
$q_\lambda, \Delta\lambda$	number of divided segments and interval such that $q_\lambda \Delta\lambda = \lambda^{max} - \lambda^{min}$
$q_{p_i}, \Delta p_i$	number of divided segments and intervals such that $q_{p_i} \Delta p_i = p_i^{max} - p_i^{min}$

Step 1: Set  $N = 0$ . Select positive scalars  $\lambda^{min}$ ,  $\lambda^{max}$ ,  $\Delta\lambda$  and  $\Delta p_i$  ( $i = 1, 2, \dots, \rho$ ) satisfying the relations defined in Table I.

Step 2: For all the combinations  $(\lambda, p_1, p_2, \dots, p_\rho)$  on all the grid points  $[\lambda^{min} \ \lambda^{max}] \times [p_1^{min} \ p_1^{max}] \times \dots \times [p_\rho^{min} \ p_\rho^{max}]$  with the intervals  $\Delta\lambda, \Delta p_1, \Delta p_2, \dots, \Delta p_\rho$ , solve

$$\min_{F_j(\mathbf{x}), \bar{g}_{ij}(\mathbf{x})} \alpha \text{ subject to (15), (16) and (17)} \quad (37)$$

and find the grid point with the minimum  $\alpha$ . If a grid point with  $\alpha < 0$  is found, it is a strict solution of Theorem 1. If any feasible solutions with  $\alpha < 0$  are not obtained, then substitute  $F_j(\mathbf{x})$ ,  $\bar{g}_{ij}(\mathbf{x})$ ,  $V(\mathbf{x})$  and  $\lambda$  obtained at the minimum grid

point into  $F_j^N(\mathbf{x})$ ,  $\bar{g}_{ij}^N(\mathbf{x})$ ,  $V^N(\mathbf{x})$  and  $\lambda^N$ , respectively, and go to Step 3.

Step 3: Set  $F_j(\mathbf{x}) = F_j^N(\mathbf{x})$ ,  $\bar{g}_{ij}(\mathbf{x}) = \bar{g}_{ij}^N(\mathbf{x})$ ,  $V(\mathbf{x}) = V^N(\mathbf{x})$  and  $\lambda = \lambda^N$ . For the given  $F_j(\mathbf{x})$ ,  $\bar{g}_{ij}(\mathbf{x})$ ,  $V(\mathbf{x})$ , and  $\lambda$ , solve the following SOS optimization problem.

$$\min_{\delta F_j(\mathbf{x}), \delta V(\mathbf{x}), \delta \bar{g}_{ij}(\mathbf{x}), \delta \lambda} \alpha \quad \text{subject to (38) } \sim (44)$$

The SOS conditions (38)  $\sim$  (44) are derived by applying the key idea to Theorem 1.

$$V(\mathbf{x}) + \delta V(\mathbf{x}) - \epsilon(\mathbf{x}) \quad \text{is SOS}, \quad (38)$$

$$\left( \sum_{k=1}^r \hat{h}_k^2 \right)^s \sum_{i=1}^r \sum_{j=1}^r \hat{h}_i^2 \hat{h}_j^2 \{ -(\bar{\Lambda}_{ij}(\mathbf{x}) + \delta \bar{\Lambda}_{ij}(\mathbf{x})) + \alpha(\delta V(\mathbf{x}) + V(\mathbf{x})) \} \quad \text{is SOS}, \quad (39)$$

$$\left( \sum_{k=1}^r \hat{h}_k^2 \right)^s \sum_{i=1}^r \sum_{j=1}^r \hat{h}_i^2 \hat{h}_j^2 \times \left[ \begin{array}{c} \lambda \bar{g}_{ij}(\mathbf{x}) + \delta \lambda \bar{g}_{ij}(\mathbf{x}) + \lambda \delta \bar{g}_{ij}(\mathbf{x}) \\ D_{ai}^T(\mathbf{x}) \boldsymbol{\mu}^T(\mathbf{x}) \\ D_{bi}^T(\mathbf{x}) \boldsymbol{\mu}^T(\mathbf{x}) \\ \beta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \\ \beta_{bi}(\mathbf{x}) \mathbf{E}_{bi}(\mathbf{x}) (F_j(\mathbf{x}) + \delta F_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) \\ * \quad * \quad * \quad * \\ 2\mathbf{I} \quad \mathbf{0} \quad \mathbf{0} \quad \mathbf{0} \\ \mathbf{0} \quad 2\mathbf{I} \quad \mathbf{0} \quad \mathbf{0} \\ \mathbf{0} \quad \mathbf{0} \quad 2\mathbf{I} \quad \mathbf{0} \\ \mathbf{0} \quad \mathbf{0} \quad \mathbf{0} \quad 2\mathbf{I} \end{array} \right] \mathbf{v}_1 \quad \text{is SOS}, \quad (40)$$

$$\mathbf{v}_2^T \left[ \begin{array}{cc} \epsilon_V V^2(\mathbf{x}) & \delta V(\mathbf{x}) \\ \delta V(\mathbf{x}) & I \end{array} \right] \mathbf{v}_2 \quad \text{is SOS}, \quad (41)$$

$$\mathbf{v}_3^T \left[ \begin{array}{cc} \epsilon_F F_j^T(\mathbf{x}) F_j(\mathbf{x}) & \delta F_j(\mathbf{x}) \\ \delta F_j^T(\mathbf{x}) & I \end{array} \right] \mathbf{v}_3 \quad \text{is SOS}, \quad (42)$$

$$\mathbf{v}_4^T \left[ \begin{array}{cc} \epsilon_g \bar{g}_{ij}^2(\mathbf{x}) & \delta \bar{g}_{ij}(\mathbf{x}) \\ \delta \bar{g}_{ij}(\mathbf{x}) & I \end{array} \right] \mathbf{v}_4 \quad \text{is SOS}, \quad (43)$$

$$\mathbf{v}_5^T \left[ \begin{array}{cc} \epsilon_L \lambda^2 & \delta \lambda \\ \delta \lambda & I \end{array} \right] \mathbf{v}_5 \quad \text{is SOS}, \quad (44)$$

where

$$\begin{aligned} \delta \bar{\Lambda}_{ij}(\mathbf{x}) &= \frac{\partial \delta V(\mathbf{x})}{\partial \mathbf{x}} \{ \mathbf{A}_i(\mathbf{x}) - \mathbf{B}_i(\mathbf{x}) F_j(\mathbf{x}) \} \hat{\mathbf{x}}(\mathbf{x}) \\ &\quad - \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \mathbf{B}_i(\mathbf{x}) \delta F_j(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) + \delta \bar{g}_{ij}(\mathbf{x}), \\ \boldsymbol{\mu}(\mathbf{x}) &= \lambda \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right) + \delta \lambda \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right) + \lambda \left( \frac{\partial \delta V(\mathbf{x})}{\partial \mathbf{x}} \right). \end{aligned}$$

$\mathbf{v}_1 - \mathbf{v}_5$  denote vectors that are independent of  $\mathbf{x}$ .  $\epsilon_V$ ,  $\epsilon_F$ ,  $\epsilon_g$  and  $\epsilon_L$  are very small positive values and  $s$  is a non-negative integer.

Step 4: For  $\delta V(\mathbf{x})$  and  $\delta \lambda$  obtained by solving the SOS optimization problem in Step 3, let  $V^{N+1}(\mathbf{x}) = V^N(\mathbf{x}) + \delta V(\mathbf{x})$  and  $\lambda^{N+1} = \lambda^N + \delta \lambda$ , respectively. Then set  $N = N + 1$ . Next, set  $V(\mathbf{x}) = V^N(\mathbf{x})$  and  $\lambda = \lambda^N$ . For the given  $V(\mathbf{x})$  and  $\lambda$ , solving the minimizing SOS problem (45).

$$\min_{F_j(\mathbf{x}), \bar{g}_{ij}(\mathbf{x})} \alpha \quad \text{subject to (15), (16) and (17)} \quad (45)$$

If a feasible solution with  $\alpha < 0$  is obtained, it is a strict solution of Theorem 1. If any feasible solutions with  $\alpha < 0$

are not obtained, then substitute  $F_j(\mathbf{x})$  and  $\bar{g}_{ij}(\mathbf{x})$  obtained by solving (45) into  $F_j^N(\mathbf{x})$  and  $\bar{g}_{ij}^N(\mathbf{x})$ , respectively, and go to Step 3.

**Remark 4.** Assume that  $\Delta_{bi}(\mathbf{x}) = \mathbf{0}$  for all  $i$ , i.e., there are no uncertainties with respect to the input terms. Then, (40) and (43) can be simplified as (46) and (47), respectively.

$$\begin{aligned} & \left( \sum_{k=1}^r \hat{h}_k^2 \right)^s \sum_{i=1}^r \sum_{j=1}^r \hat{h}_i^2 \hat{h}_j^2 \times \\ & \mathbf{v}_1^T \left[ \begin{array}{c} \lambda \bar{g}_i(\mathbf{x}) + \delta \lambda \bar{g}_i(\mathbf{x}) + \lambda \delta \bar{g}_i(\mathbf{x}) \\ D_{ai}^T(\mathbf{x}) \boldsymbol{\mu}^T(\mathbf{x}) \\ \beta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \\ * \quad * \\ 2\mathbf{I} \quad \mathbf{0} \\ \mathbf{0} \quad 2\mathbf{I} \end{array} \right] \mathbf{v}_1 \quad \text{is SOS}, \quad (46) \\ & \mathbf{v}_4^T \left[ \begin{array}{cc} \epsilon_g \bar{g}_i^2(\mathbf{x}) & \delta \bar{g}_i(\mathbf{x}) \\ \delta \bar{g}_i(\mathbf{x}) & I \end{array} \right] \mathbf{v}_4 \quad \text{is SOS}. \quad (47) \end{aligned}$$

**Remark 5.** We need to carefully deal with SOS solutions since some numerical reliability options exist in the SOS solvers and their feasible results might be changed very slightly according to the options, particularly, for complicated systems. In other words, feasible area plots (, e.g., such as Figs. 4 and 5) might change very slightly according to the options. To obtain more reliable solutions for SOS conditions, we perform the following double checking throughout this paper. After getting a feasible solution in the algorithm, we carefully perform the so-called SOS test (, e.g., 'issos' command in SOSOPT) for the polynomials calculated by substituting the feasible solution into the considered SOS conditions. That is, with one of most reliable options, we check whether the polynomials (calculated by substituting the feasible solution into the considered SOS conditions) are judged as SOS polynomials or not. If the check returns an infeasible result, we strictly judge 'infeasible'. This double checking is important to have reliable solutions in the use of SOSOPT[38] or SOSTOOLS [39] and an SDP solver [40], [41]

## V. DESIGN EXAMPLES

### A. Design Example 1

Consider the following nonlinear system with an uncertainty.

$$\begin{aligned} \dot{x}_1 &= (-1 + \Delta(t) + x_1 + x_1^2 + x_1 x_2 - x_2^2) x_1 \\ &\quad + x_2 + x_1 u, \\ \dot{x}_2 &= -2 \sin(x_1) - 6x_2 + 7u, \end{aligned} \quad (48)$$

where  $\Delta(t)$  is the uncertainty satisfying  $|\Delta(t)| \leq c$  for all  $t$ . All the simulation results given in this design example are carried out for  $\Delta(t) = c \sin(200\pi t)$ . However, it should be noted that  $\Delta(t)$  is the uncertain term and only its upper bound, i.e.  $c$ , is known as well as the standard robust control setting.

Using the sector nonlinearity technique [2], the nonlinear system with the uncertainty is exactly converted into the following two-rule polynomial fuzzy system with uncertainties:

$$\dot{\mathbf{x}} = \sum_{i=1}^r h_i(z) \{ (\mathbf{A}_i(\mathbf{x}) + \mathbf{D}_{ai}(\mathbf{x}) \boldsymbol{\Delta}_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x})) \mathbf{x} + \mathbf{B}_i(\mathbf{x}) \mathbf{u} \},$$

where  $r=2$ ,  $\hat{\mathbf{x}}(\mathbf{x}) = \mathbf{x} = [x_1, x_2]$ ,  $\mathbf{z} = x_1$ , and

$$\begin{aligned} \mathbf{A}_1(\mathbf{x}) &= \begin{bmatrix} -1 + x_1 + x_1^2 + x_1x_2 - x_2^2 & 1 \\ -2 & -6 \end{bmatrix}, \\ \mathbf{A}_2(\mathbf{x}) &= \begin{bmatrix} -1 + x_1 + x_1^2 + x_1x_2 - x_2^2 & 1 \\ 0.4344 & -6 \end{bmatrix}, \\ \mathbf{B}_1(\mathbf{x}) &= \mathbf{B}_2(\mathbf{x}) = \begin{bmatrix} x_1 \\ 7 \end{bmatrix}, \\ \mathbf{D}_{a1}(\mathbf{x}) &= \mathbf{D}_{a2}(\mathbf{x}) = \begin{bmatrix} c \\ 0 \end{bmatrix}, \\ \Delta_{a1}(\mathbf{x}) &= \Delta_{a2}(\mathbf{x}) = \Delta(t)/c, \\ \mathbf{E}_{a1}(\mathbf{x}) &= \mathbf{E}_{a2}(\mathbf{x}) = \begin{bmatrix} 1 & 0 \end{bmatrix}, \\ h_1(\mathbf{z}) &= \frac{\sin(x_1) + 0.2172x_1}{1.2172x_1}, \quad h_2(\mathbf{z}) = \frac{x_1 - \sin(x_1)}{1.2172x_1}. \end{aligned}$$

Since  $\|\Delta_{a1}(\mathbf{x})\| = \|\Delta_{a2}(\mathbf{x})\| = \|\Delta(t)/c\| \leq 1$ , we have  $\beta_{a1} = \beta_{a2} = 1$ . Moreover, the algorithm presented in Section IV is carried out with the initial setting  $s = 0$ ,  $\epsilon_g = 0.001$ ,  $\epsilon_F = 0.001$ ,  $\epsilon_V = 0.001$ ,  $\epsilon_L = 0.001$ ,  $\lambda^{min} = 0.2$ ,  $\lambda^{max} = 5$ ,  $\Delta\lambda = 0.8$ ,  $p_i^{min} = 0.2$ ,  $p_i^{max} = 1$ ,  $\Delta p_i = 0.2$  for  $i = 1, 2$ . To show the validity of derived conditions, we compare the feasible values of  $c$  for the proposed robust control design method and the SOS-based design method of [36]. The proposed robust control design method is feasible for  $c \leq 0.76$ , and the method of [36] is feasible for  $c \leq 0.39$ . It shows that the proposed robust design method provides more relaxed results than the method of [36].

TABLE II  
FEASIBLE AREAS FOR  $c$ .

Convex SOS robust [36]	$c \leq 0.39$
Our SOS robust	$c \leq 0.76$

For  $c = 0.76$ , Fig. 2 shows the behavior of the nonlinear system (48) with  $u = 0$ . Thus, the system is unstable when  $u = 0$ . By solving the conditions in Theorem 2, a feasible solution for  $c = 0.76$  can be obtained as

$$\begin{aligned} \lambda &= 0.6811, \\ V(\mathbf{x}) &= 1.0524x_1^2 + 0.1361x_2^2 \\ F_1(\mathbf{x}) &= \begin{bmatrix} 1.6566x_1 + 0.2669x_2 + 0.8952 \\ 0.2669x_1 - 0.1902 \end{bmatrix}^T, \\ F_2(\mathbf{x}) &= \begin{bmatrix} 1.6314x_1 + 0.2696x_2 + 1.0418 \\ 0.2696x_1 - 0.2061 \end{bmatrix}^T, \\ \bar{g}_1(\mathbf{x}) &= 0.6850x_1^4 - 0.5481x_1^3x_2 + 1.1416x_1^2x_2^2 \\ &\quad - 0.0932x_1^3 + 1.4081x_1^2x_2 + 0.5751x_1x_2^2 \\ &\quad + 1.8407x_1^2 + 0.0615x_1x_2 + 0.6297x_2^2, \\ \bar{g}_2(\mathbf{x}) &= 0.6797x_1^4 - 0.5397x_1^3x_2 + 1.1377x_1^2x_2^2 \\ &\quad + 0.0217x_1^3 + 1.3704x_1^2x_2 + 0.5679x_1x_2^2 \\ &\quad + 1.8395x_1^2 - 0.0914x_1x_2 + 0.6296x_2^2. \end{aligned}$$

Fig. 3 shows the controlled behavior for six different initial conditions. It can be seen from Fig. 3 that the design fuzzy controller stabilizes the system from all the initial conditions although the system has uncertainties.

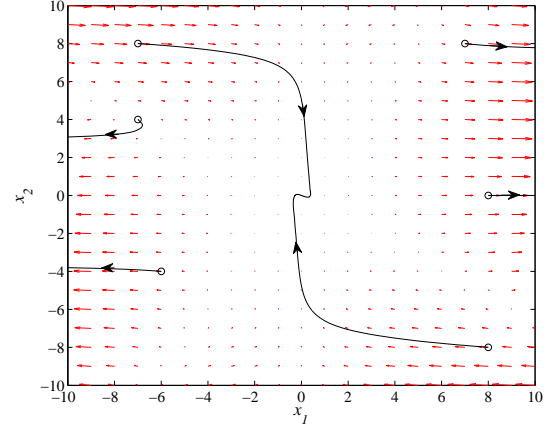


Fig. 2. Behavior of the nonlinear system (48) with  $u = 0$ .

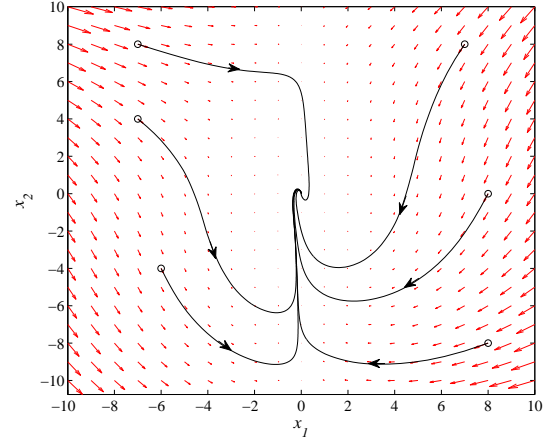


Fig. 3. Controlled behavior of the nonlinear system (48).

Based on the sector nonlinearity technique [2], the nonlinear system (48) can be exactly represented by a T-S fuzzy model for  $x_1 \in [-d_1, d_1]$  and  $x_2 \in [-d_2, d_2]$ , where  $d_1$  and  $d_2$  are constant satisfying  $0 < d_1 < \infty$  and  $0 < d_2 < \infty$ . The T-S fuzzy model is obtained as

$$\sum_{i=1}^8 h_i(\mathbf{z}) \{ (\mathbf{A}_i + \mathbf{D}_{ai} \Delta_{ai}(t) \mathbf{E}_{ai}) \mathbf{x} + \mathbf{B}_i \mathbf{u} \}, \quad (49)$$

where

$$\begin{aligned} \mathbf{A}_1 &= \begin{bmatrix} k_{max} & 1 \\ -2 & -6 \end{bmatrix}, \quad \mathbf{A}_2 = \begin{bmatrix} k_{max} & 1 \\ -2 & -6 \end{bmatrix}, \\ \mathbf{A}_3 &= \begin{bmatrix} k_{max} & 1 \\ -2 \sin(d_1) & -6 \end{bmatrix}, \quad \mathbf{A}_4 = \begin{bmatrix} k_{max} & 1 \\ -2 \sin(d_1) & -6 \end{bmatrix}, \\ \mathbf{A}_5 &= \begin{bmatrix} k_{min} & 1 \\ -2 & -6 \end{bmatrix}, \quad \mathbf{A}_6 = \begin{bmatrix} k_{min} & 1 \\ -2 & -6 \end{bmatrix}, \\ \mathbf{A}_7 &= \begin{bmatrix} k_{min} & 1 \\ -2 \sin(d_1) & -6 \end{bmatrix}, \quad \mathbf{A}_8 = \begin{bmatrix} k_{min} & 1 \\ -2 \sin(d_1) & -6 \end{bmatrix}, \end{aligned}$$



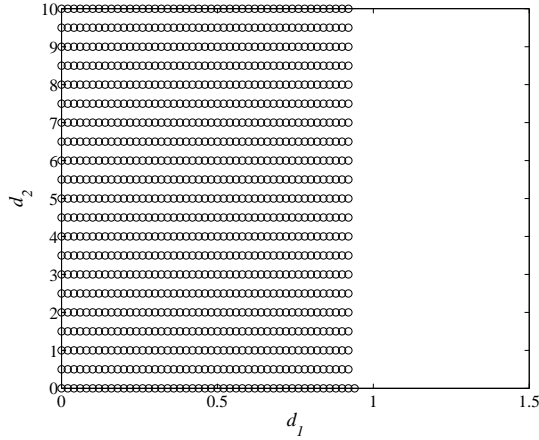


Fig. 4. Feasible area of the LMI-based robust control design conditions proposed in [2] for the T-S fuzzy model (49) with  $c = 0.76$ .

$$\begin{aligned}
 B_1 = B_3 = B_5 = B_7 &= \begin{bmatrix} d_1 \\ 7 \end{bmatrix}, \\
 B_2 = B_4 = B_6 = B_8 &= \begin{bmatrix} -d_1 \\ 7 \end{bmatrix}, \\
 D_{ai} &= \begin{bmatrix} c \\ 0 \end{bmatrix}, \quad i = 1, \dots, 8, \\
 \Delta_{ai} &= \Delta(t)/c, \quad i = 1, \dots, 8, \\
 E_{ai} &= \begin{bmatrix} 1 & 0 \end{bmatrix}, \quad i = 1, \dots, 8, \\
 k_{max} &= \max_{|x_1| < d_1, |x_2| < d_2} (-1 + x_1 + x_1^2 + x_1x_2 - x_2^2), \\
 k_{min} &= \min_{|x_1| < d_1, |x_2| < d_2} (-1 + x_1 + x_1^2 + x_1x_2 - x_2^2).
 \end{aligned}$$

The membership functions are given as follows.

$$\begin{aligned}
 h_1(z) &= \frac{k - k_{min}}{k_{max} - k_{min}} \cdot \frac{\sin x_1 - (\sin d_1/d_1)x_1}{(1 - (\sin d_1/d_1))x_1} \cdot \frac{x_1 + d_1}{2d_1} \\
 h_2(z) &= \frac{k - k_{min}}{k_{max} - k_{min}} \cdot \frac{\sin x_1 - (\sin d_1/d_1)x_1}{(1 - (\sin d_1/d_1))x_1} \cdot \frac{d_1 - x_1}{2d_1} \\
 h_3(z) &= \frac{k - k_{min}}{k_{max} - k_{min}} \cdot \frac{x_1 - \sin x_1}{(1 - (\sin d_1/d_1))x_1} \cdot \frac{x_1 + d_1}{2d_1} \\
 h_4(z) &= \frac{k - k_{min}}{k_{max} - k_{min}} \cdot \frac{x_1 - \sin x_1}{(1 - (\sin d_1/d_1))x_1} \cdot \frac{d_1 - x_1}{2d_1} \\
 h_5(z) &= \frac{k_{max} - k}{k_{max} - k_{min}} \cdot \frac{\sin x_1 - (\sin d_1/d_1)x_1}{(1 - (\sin d_1/d_1))x_1} \cdot \frac{x_1 + d_1}{2d_1} \\
 h_6(z) &= \frac{k_{max} - k}{k_{max} - k_{min}} \cdot \frac{\sin x_1 - (\sin d_1/d_1)x_1}{(1 - (\sin d_1/d_1))x_1} \cdot \frac{d_1 - x_1}{2d_1} \\
 h_7(z) &= \frac{k_{max} - k}{k_{max} - k_{min}} \cdot \frac{x_1 - \sin x_1}{(1 - (\sin d_1/d_1))x_1} \cdot \frac{x_1 + d_1}{2d_1} \\
 h_8(z) &= \frac{k_{max} - k}{k_{max} - k_{min}} \cdot \frac{x_1 - \sin x_1}{(1 - (\sin d_1/d_1))x_1} \cdot \frac{d_1 - x_1}{2d_1}
 \end{aligned}$$

Fig. 4 shows the feasible area of the LMI-based robust control design conditions proposed in [2] for the T-S fuzzy model (49) with  $c = 0.76$ .

**Remark 6.** For nonlinear systems with polynomial terms, it is impossible to exactly construct a global T-S fuzzy model. In this example, a local T-S fuzzy model with 8 rules can

be constructed by assuming the ranges of  $x_1$  and  $x_2$ , e.g.,  $|x_1| < d_1$  and  $|x_2| < d_2$ , where  $d_1$  and  $d_2$  are nonnegative values. If we select huge values for  $d_1$  and  $d_2$ , the local T-S fuzzy model could be a global model, however, it becomes much harder to guarantee the stability for larger values of  $d_1$  and  $d_2$ . In other words, smaller values of  $d_1$  and  $d_2$  becomes easier to guarantee the stability of the local T-S fuzzy model. However, the LMI robust conditions for T-S fuzzy models are infeasible even for very small values, e.g.,  $d_1 > 0.96$  when  $c = 0.76$ .

## B. Design example II

Consider the following nonlinear system with uncertainties.

$$\begin{aligned}
 \dot{x}_1 &= (-1 + \Delta_a(t) + x_1 + x_1^2 + x_1x_2 - x_2^2)x_1 \\
 &\quad + x_2 + x_1u, \\
 \dot{x}_2 &= -2 \sin(x_1)x_1 - 6x_2 - 4 \sin(x_1)(1 + \Delta_b(t))u,
 \end{aligned} \tag{50}$$

where  $\Delta_a(t)$  and  $\Delta_b(t)$  are the uncertainties satisfying  $|\Delta_a(t)| \leq c_a$  and  $|\Delta_b(t)| \leq c_b$  for all  $t$ . All the simulation results given in this design example are carried out for  $\Delta_a(t) = c_a \sin(200\pi t)$  and  $\Delta_b(t) = c_b \sin(200\pi t)$ . However, it should be noted that  $\Delta_a(t)$  and  $\Delta_b(t)$  are the uncertain terms and only their upper bounds, i.e.  $c_a$  and  $c_b$ , are known as well as the standard robust control setting.

Using the sector nonlinearity technique [2], the nonlinear system with the uncertainties is exactly converted into the polynomial fuzzy system (9) with  $r=2$ ,  $\hat{x}(x) = x = [x_1, x_2]$ ,  $z = x_1$  and

$$\begin{aligned}
 A_1(x) &= \begin{bmatrix} -1 + x_1 + x_1^2 + x_1x_2 - x_2^2 & 1 \\ -2 & -6 \end{bmatrix}, \\
 A_2(x) &= \begin{bmatrix} -1 + x_1 + x_1^2 + x_1x_2 - x_2^2 & 1 \\ 2 & -6 \end{bmatrix}, \\
 B_1(x) &= \begin{bmatrix} x_1 \\ -4 \end{bmatrix}, \quad B_2(x) = \begin{bmatrix} x_1 \\ 4 \end{bmatrix}, \\
 D_{a1}(x) &= D_{a2}(x) = \begin{bmatrix} c_a \\ 0 \end{bmatrix}, \\
 \Delta_{a1}(x) &= \Delta_{a2}(x) = \Delta_a(t)/c_a, \\
 E_{a1}(x) &= E_{a2}(x) = \begin{bmatrix} 1 & 0 \end{bmatrix} \\
 D_{b1}(x) &= \begin{bmatrix} 0 \\ -4c_b \end{bmatrix}, \quad D_{b2}(x) = \begin{bmatrix} 0 \\ 4c_b \end{bmatrix}, \\
 \Delta_{b1}(x) &= \Delta_{b2}(x) = \Delta_b(t)/c_b, \\
 E_{b1}(x) &= E_{b2}(x) = 1, \\
 h_1(z) &= \frac{\sin(x_1) + 1}{2}, \quad h_2(z) = \frac{1 - \sin(x_1)}{2}.
 \end{aligned}$$

Since  $\|\Delta_{a1}(x)\| = \|\Delta_{a2}(x)\| = \|\Delta_a(t)/c_a\| \leq 1$  and  $\|\Delta_{b1}(x)\| = \|\Delta_{b2}(x)\| = \|\Delta_b(t)/c_b\| \leq 1$ , we have  $\beta_{a1} = \beta_{a2} = \beta_{b1} = \beta_{b2} = 1$ . Moreover, the algorithm presented in Section IV is carried out with the initial setting  $s = 0$ ,  $\epsilon_g = 0.001$ ,  $\epsilon_F = 0.001$ ,  $\epsilon_V = 0.001$ ,  $\epsilon_L = 0.001$ ,  $\lambda^{min} = 0.2$ ,  $\lambda^{max} = 5$ ,  $\Delta\lambda = 0.8$ ,  $p_i^{min} = 0.2$ ,  $p_i^{max} = 1$ ,  $\Delta p_i = 0.2$  for  $i = 1, 2$ . To show the validity of derived conditions, we compare the feasible areas in the region ( $0.01 \leq c_a \leq 0.12$  and  $0.01 \leq c_b \leq 0.26$ ) for the proposed robust control design method and the SOS-based design method of [36] as shown in Fig. 5. It shows that the proposed robust

design method provides more relaxed results than the method of [36].

For  $c_a = c_b = 0.1$ , Fig. 6 shows the behavior of the nonlinear system (50) with  $u = 0$ . By solving the conditions in Theorem 1, a feasible solution for  $c_a = c_b = 0.1$  can be obtained as

$$\begin{aligned} \lambda &= 6.806, \\ V(\mathbf{x}) &= 1.3643x_1^2 + 0.1451x_2^2 \\ F_1(\mathbf{x}) &= \begin{bmatrix} 1.6015x_1 + 0.3447x_2 + 0.7279 \\ 0.3447x_1 + 0.0207 \end{bmatrix}^T, \\ F_2(\mathbf{x}) &= \begin{bmatrix} 1.6672x_1 + 0.3119x_2 + 0.7391 \\ 0.3119x_1 - 0.0983 \end{bmatrix}^T, \\ \bar{g}_{1,1}(\mathbf{x}) &= 0.4911x_1^4 + 0.0476x_1^3x_2 + 0.3544x_1^2x_2^2 \\ &\quad + 0.0276x_1^3 - 0.1997x_1^2x_2 - 0.1470x_1x_2^2 \\ &\quad + 0.6121x_1^2 - 0.2999x_1x_2 + 0.2630x_2^2, \\ \bar{g}_{1,2}(\mathbf{x}) &= 0.5928x_1^4 + 0.1080x_1^3x_2 + 0.4294x_1^2x_2^2 \\ &\quad + 0.1331x_1^3 - 0.0122x_1^2x_2 - 0.0341x_1x_2^2 \\ &\quad + 0.7168x_1^2 - 0.1211x_1x_2 + 0.4064x_2^2, \\ \bar{g}_{2,1}(\mathbf{x}) &= 0.5621x_1^4 + 0.1221x_1^3x_2 + 0.4267x_1^2x_2^2 \\ &\quad + 0.1256x_1^3 + 0.0282x_1^2x_2 - 0.0193x_1x_2^2 \\ &\quad + 0.7251x_1^2 - 0.1052x_1x_2 + 0.3879x_2^2, \\ \bar{g}_{2,2}(\mathbf{x}) &= 0.5316x_1^4 + 0.0840x_1^3x_2 + 0.3902x_1^2x_2^2 \\ &\quad + 0.1579x_1^3 + 0.1456x_1^2x_2 + 0.0557x_1x_2^2 \\ &\quad + 0.6790x_1^2 - 0.1412x_1x_2 + 0.3752x_2^2. \end{aligned}$$

Fig. 7 shows the controlled behavior for six different initial conditions. It can be seen from Fig. 7 that the design fuzzy controller stabilizes the system from all the initial conditions although the system has uncertainties.

**Remark 7.** *Design Examples I and II show that our approach provides more relaxed results than the existing LMI approach and the existing SOS approach. In addition, as mentioned in*

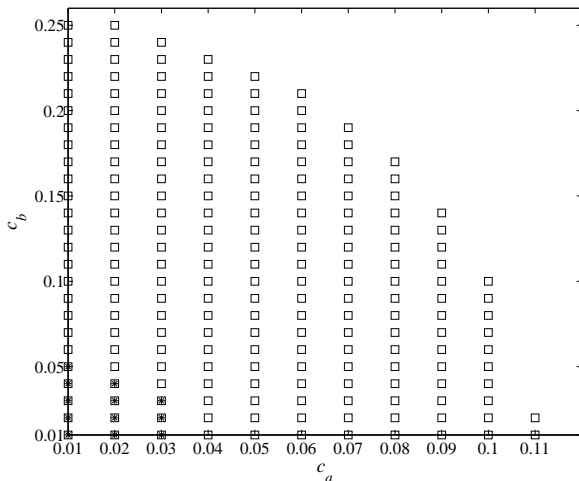


Fig. 5. Feasible areas for proposed robust control design method ( $\square$ ) and the SOS-based design method of [36] (\*).

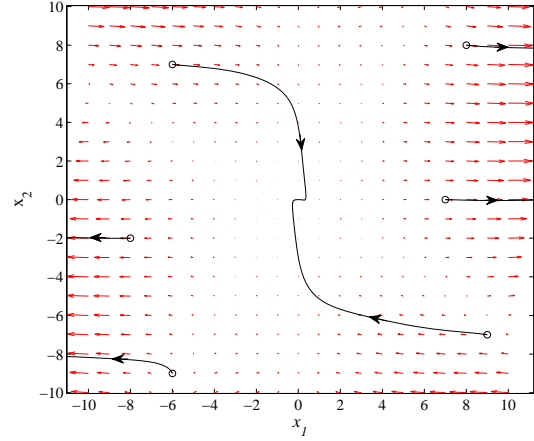


Fig. 6. Behavior of the nonlinear system (50) with  $u = 0$ .

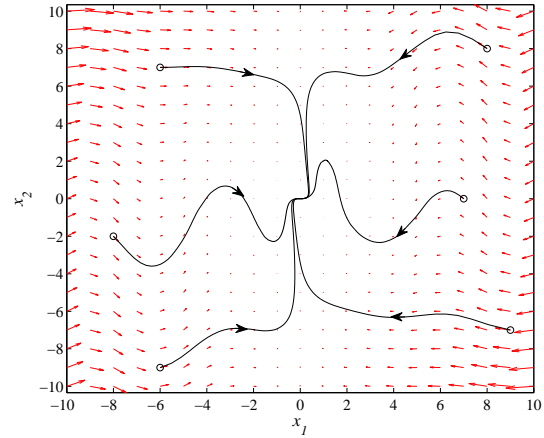


Fig. 7. Controlled behavior of the nonlinear system (50).

*Remark 6, the LMI-based approach for the T-S fuzzy model does not guarantee the global stability of the nonlinear system.*

## VI. SEMI-GLOBAL ROBUST STABILIZATION CONDITIONS WITH CONSIDERING INPUT CONSTRAINTS

Sections III and V gave global robust stabilization conditions and their design examples. It is known that the global stabilization is sometimes difficult to be achieved for complicated systems, e.g., unmanned aerial vehicles (UAVs), in practical. Moreover, it is usually the case that the input constraints exist in practical systems. Therefore, Section VI proposes a semi-global robust control design method with considering the input constraints. Section VII will show altitude control of a paraglider-type UAV as a design example of the semi-global robust stabilization with considering the input constraints.

Consider the operation domain

$$D_o = \{\mathbf{x} : x_\beta^{\min} \leq x_\beta \leq x_\beta^{\max}, \beta = 1, \dots, n\} \quad (51)$$

and input constraints

$$u_\ell^{\min} \leq u_\ell \leq u_\ell^{\max}, \ell = 1, \dots, m. \quad (52)$$

For the operation domain (51), the semi-global robust control satisfying the input constrains (52) can be designed by the following theorem.

**Theorem 3.** *The polynomial controller (10) satisfying the input constraints (52) stabilizes the system (9) and the outmost Lyapunov function level set  $\Omega_{V,\gamma} = \{\mathbf{x} : V(\mathbf{x}) \leq \gamma\}$  contained in the operation domain (51) is a contractively invariant set if there exist a polynomial function  $V(\mathbf{x})$ , polynomial matrices  $\mathbf{F}_j(\mathbf{x})$ , polynomials  $\bar{g}_{ij}(\mathbf{x})$ ,  $Q_\beta(\mathbf{x})$ ,  $\tau_\beta(\mathbf{x})$ ,  $\varphi_\beta(\mathbf{x})$  and a scalar  $\alpha < 0$  such that (17) and the following conditions hold:*

$$V(\mathbf{x}) - \epsilon(\mathbf{x}) \text{ is SOS,} \quad (53)$$

$$\left(\sum_{k=1}^r \hat{h}_k^2\right)^s \sum_{i=1}^r \sum_{j=1}^r \hat{h}_i^2 \hat{h}_j^2 \{-\Upsilon_{ij}(\mathbf{x}) + \alpha V(\mathbf{x})\} \text{ is SOS,} \quad (54)$$

$$Q_\beta(\mathbf{x}) \text{ is SOS, } \beta = 1, \dots, n, \quad (55)$$

$$\sum_{\beta=1}^n \varphi_\beta(\mathbf{x})(x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}) - \mathbf{d}_\ell \mathbf{F}_i(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) - u_\ell^{\min} \text{ is SOS, } i = 1, \dots, r, \ell = 1, \dots, m, \quad (56)$$

$$\sum_{\beta=1}^n \tau_\beta(\mathbf{x})(x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}) + \mathbf{d}_\ell \mathbf{F}_i(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) + u_\ell^{\max} \text{ is SOS, } i = 1, \dots, r, \ell = 1, \dots, m, \quad (57)$$

$$\varphi_\beta(\mathbf{x}) \text{ is SOS, } \beta = 1, \dots, n, \quad (58)$$

$$\tau_\beta(\mathbf{x}) \text{ is SOS, } \beta = 1, \dots, n, \quad (59)$$

where  $\epsilon(\mathbf{x})$  is a radially unbounded positive definite polynomial.  $\mathbf{d}_\ell = [d_1^\ell \ d_2^\ell \ \dots \ d_m^\ell]$  with  $d_\ell^\ell = 1$  and  $d_\ell^j = 0 \ \forall j \neq \ell$ .  $s$  is a nonnegative integer, and

$$\begin{aligned} \Upsilon_{ij}(\mathbf{x}) &= \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) + \bar{g}_{ij}(\mathbf{x}) \\ &\quad - \sum_{\beta=1}^n Q_\beta(\mathbf{x})(x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}). \end{aligned}$$

If a solution satisfying the conditions (17), (53)~(59) is found, the outmost Lyapunov function level set  $\Omega_{V,\gamma} = \{\mathbf{x} : V(\mathbf{x}) \leq \gamma\}$  contained in the operation domain (51), i.e. the contractively invariant set, can be obtained by solving the following optimization problem.

$$\begin{aligned} \max_{\phi_\beta(\mathbf{x})} \quad & \gamma \text{ subject to} \\ & \phi_\beta(\mathbf{x})(V(\mathbf{x}) - \gamma) - (x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}) \\ & \text{is SOS, } \beta = 1, \dots, n, \quad (60) \end{aligned}$$

$$\phi_\beta(\mathbf{x}) \text{ is SOS, } \beta = 1, \dots, n. \quad (61)$$

*Proof:* In the proof, we need to show that

1) If the conditions (17), (53)~(55) hold, then the outmost Lyapunov function level set  $\Omega_{V,\gamma} = \{\mathbf{x} : V(\mathbf{x}) \leq \gamma\}$  contained in the operation domain (51) is a contractively invariant set;

2) If the conditions (56)~(59) hold, then the input constraints (52) are satisfied in the operation domain (51).

1) For the operation domain (51), the following condition holds:

$$\sum_{\beta=1}^n Q_\beta(\mathbf{x})(x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}) \leq 0 \quad (62)$$

where  $Q_\beta(\mathbf{x}) \geq 0$  that is guaranteed by (55). From (21) and (22), if (17) holds, then

$$\begin{aligned} \dot{V}(\mathbf{x}) &\leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left\{ \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) \right. \\ &\quad \left. - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) + \bar{g}_{ij}(\mathbf{x}) \right\}. \quad (63) \end{aligned}$$

It can be obtained from (62) and (63) that  $\dot{V}(\mathbf{x}) \leq \alpha V(\mathbf{x}) < 0$  for  $D_o - \{0\}$  if there exist  $\alpha < 0$  such that

$$\begin{aligned} \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left\{ \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) \right. \\ \left. + \bar{g}_{ij}(\mathbf{x}) - \alpha V(\mathbf{x}) \right. \\ \left. - \sum_{\beta=1}^n Q_\beta(\mathbf{x})(x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}) \right\} \leq 0. \quad (64) \end{aligned}$$

By applying the copositivity presented in Corollary 1, the condition (64) holds if (54) is satisfied. Furthermore, if (53) holds, then  $V(\mathbf{x})$  is a positive definite and radially unbounded function which means that the level set  $\Omega_{V,\gamma}$  is bounded for any value of  $\gamma > 0$ . Consequently, if the conditions (17), (53)~(55) hold, then the outmost Lyapunov function level set  $\Omega_{V,\gamma}$  contained in the operation domain (51) is a contractively invariant set. Moreover, by applying polynomial S-procedure,  $\Omega_v \subseteq D_o$  is carried out by (60) and (61).

2) For the operation domain (51), the following two conditions hold:

$$-\sum_{\beta=1}^n \varphi_\beta(\mathbf{x})(x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}) \geq 0, \quad (65)$$

$$-\sum_{\beta=1}^n \tau_\beta(\mathbf{x})(x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}) \geq 0, \quad (66)$$

where  $\varphi_\beta(\mathbf{x}) \geq 0$  and  $\tau_\beta(\mathbf{x}) \geq 0$  that are guaranteed by (58) and (59) respectively. By applying the vector  $\mathbf{d}_\ell$ , the  $l$ -th input can be represented as

$$u_\ell = \mathbf{d}_\ell \mathbf{u} = -\sum_{i=1}^r h_i \mathbf{d}_\ell \mathbf{F}_i(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}). \quad (67)$$

It can be obtained from (65) and (67) that  $u_\ell - u_\ell^{\min} \geq 0$  for the operation domain (51) if the following condition holds:

$$\begin{aligned} -\sum_{i=1}^r h_i \mathbf{d}_\ell \mathbf{F}_i(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) - u_\ell^{\min} \\ \geq -\sum_{\beta=1}^n \varphi_\beta(\mathbf{x})(x_\beta - x_\beta^{\min})(x_\beta - x_\beta^{\max}). \quad (68) \end{aligned}$$

It is obviously that (68) holds if (56) is satisfied. On the other hand, it can be obtained from (66) and (67) that  $u_\ell^{\max} - u_\ell \geq 0$

for the operation domain (51) if the following condition holds:

$$\begin{aligned} u_\ell^{max} + \sum_{i=1}^r h_i \mathbf{d}_i \mathbf{F}_i(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) \\ \geq - \sum_{\beta=1}^n \tau_\beta(\mathbf{x})(x_\beta - x_\beta^{min})(x_\beta - x_\beta^{max}). \end{aligned} \quad (69)$$

It is obviously that (69) holds if (57) is satisfied. ■

**Theorem 4.** Assume that  $\Delta_{bi}(\mathbf{x}) = 0 \forall i$ , i.e. there are no uncertainties with respect to input terms. The polynomial controller (10) satisfying the input constraints (52) stabilizes the system (9) and the outmost Lyapunov function level set  $\Omega_{V,\gamma} = \{\mathbf{x} : V(\mathbf{x}) \leq \gamma\}$  contained in the operation domain (51) is a contractively invariant set if there exist a polynomial function  $V(\mathbf{x})$ , polynomial matrices  $\mathbf{F}_j(\mathbf{x})$ , polynomials  $\bar{g}_i(\mathbf{x})$ ,  $Q_\beta(\mathbf{x})$ ,  $\tau_\beta(\mathbf{x})$ ,  $\varphi_\beta(\mathbf{x})$  and a scalar  $\alpha < 0$  such that (55)~(59) and the following conditions hold:

$$V(\mathbf{x}) - \epsilon(\mathbf{x}) \text{ is SOS,} \quad (70)$$

$$\left( \sum_{k=1}^r \hat{h}_k^2 \right)^s \sum_{i=1}^r \sum_{j=1}^r \hat{h}_i^2 \hat{h}_j^2 \{-\bar{\Upsilon}_{ij}(\mathbf{x}) + \alpha V(\mathbf{x})\} \text{ is SOS,} \quad (71)$$

$$\mathbf{v}_1^T \begin{bmatrix} \lambda \bar{g}_i(\mathbf{x}) & * & * \\ \lambda \mathbf{D}_{ai}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T & 2\mathbf{I} & \mathbf{0} \\ \beta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) & \mathbf{0} & 2\mathbf{I} \end{bmatrix} \mathbf{v}_1 \text{ is SOS,} \quad (72)$$

where  $\epsilon(\mathbf{x})$  is a radially unbounded positive definite polynomial.  $s$  is a nonnegative integer, and

$$\begin{aligned} \bar{\Upsilon}_{ij}(\mathbf{x}) = \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) - \mathbf{B}_i(\mathbf{x}) \mathbf{F}_j(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) + \bar{g}_i(\mathbf{x}) \\ - \sum_{\beta=1}^n Q_\beta(\mathbf{x})(x_\beta - x_\beta^{min})(x_\beta - x_\beta^{max}). \end{aligned}$$

*Proof:* The proof is omitted since it is directly obtained from Theorem 3. In this case, (17) is reduced to (72). ■

**Theorem 5.** Assume that  $\mathbf{B}_i(\mathbf{x}) = \mathbf{B}(\mathbf{x})$ ,  $\mathbf{D}_{bi}(\mathbf{x}) = \mathbf{D}_b(\mathbf{x})$ ,  $\Delta_{bi}(\mathbf{x}) = \Delta_b(\mathbf{x})$  and  $\mathbf{E}_{bi}(\mathbf{x}) = \mathbf{E}_b(\mathbf{x})$  for all  $i$ . The polynomial controller (10) satisfying the input constraints (52) stabilizes the system (9) and the outmost Lyapunov function level set  $\Omega_{V,\gamma} = \{\mathbf{x} : V(\mathbf{x}) \leq \gamma\}$  contained in the operation domain (51) is a contractively invariant set if there exist a polynomial function  $V(\mathbf{x})$ , polynomial matrices  $\mathbf{F}_i(\mathbf{x})$ , polynomials  $\bar{g}_i(\mathbf{x})$ ,  $Q_\beta(\mathbf{x})$ ,  $\tau_\beta(\mathbf{x})$ ,  $\varphi_\beta(\mathbf{x})$  and a scalar  $\alpha < 0$  such that (55)~(59) and the following conditions hold:

$$V(\mathbf{x}) - \epsilon(\mathbf{x}) \text{ is SOS,} \quad (73)$$

$$\sum_{i=1}^r \hat{h}_i^2 \{-\bar{\Upsilon}_i(\mathbf{x}) + \alpha V(\mathbf{x})\} \text{ is SOS,} \quad (74)$$

$$\mathbf{v}_1^T \mathbf{L}_i(\lambda, \mathbf{x}) \mathbf{v}_1 \text{ is SOS,} \quad (75)$$

where  $\epsilon(\mathbf{x})$  is a radially unbounded positive definite poly-

nomial.  $s$  is a nonnegative integer, and

$$\begin{aligned} \bar{\Upsilon}_i(\mathbf{x}) = \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} (\mathbf{A}_i(\mathbf{x}) - \mathbf{B}(\mathbf{x}) \mathbf{F}_i(\mathbf{x})) \hat{\mathbf{x}}(\mathbf{x}) + \bar{g}_i(\mathbf{x}), \\ - \sum_{\beta=1}^n Q_\beta(\mathbf{x})(x_\beta - x_\beta^{min})(x_\beta - x_\beta^{max}), \end{aligned}$$

$$\mathbf{L}_i(\lambda, \mathbf{x}) =$$

$$\begin{bmatrix} \lambda \bar{g}_i(\mathbf{x}) & * & * & * & * \\ \lambda \mathbf{D}_{ai}^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T & 2\mathbf{I} & \mathbf{0} & \mathbf{0} & \mathbf{0} \\ \lambda \mathbf{D}_b^T(\mathbf{x}) \left( \frac{\partial V(\mathbf{x})}{\partial \mathbf{x}} \right)^T & \mathbf{0} & 2\mathbf{I} & \mathbf{0} & \mathbf{0} \\ \beta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) & \mathbf{0} & \mathbf{0} & 2\mathbf{I} & \mathbf{0} \\ \beta_b(\mathbf{x}) \mathbf{E}_b(\mathbf{x}) \mathbf{F}_i(\mathbf{x}) \hat{\mathbf{x}}(\mathbf{x}) & \mathbf{0} & \mathbf{0} & \mathbf{0} & 2\mathbf{I} \end{bmatrix}.$$

*Proof:* The proof is omitted since it is directly obtained by the same fashion as in the proof of Theorem 3. ■

## VII. DESIGN EXAMPLE OF SEMI-GLOBAL ROBUST STABILIZATION WITH CONSIDERING INPUT CONSTRAINTS

Section VII gives a design example of the semi-global robust stabilization for a paraglider-type UAV. Under some assumptions, the altitude error dynamics of a paraglider-type UAV around the trimmed equilibrium are given as

$$\begin{aligned} \ddot{e}(t) = & 0.1336(1 + \Delta_\theta(t))u(t) \\ & + p_1(\dot{e}(t))(1 + \Delta_\alpha(t)) \cos(0.2\dot{e}(t))\dot{e}(t) \\ & + p_2(\dot{e}(t))(1 + \Delta_\alpha(t)) \sin(0.2\dot{e}(t))\dot{e}(t), \end{aligned} \quad (76)$$

where  $e(t)$  denotes the altitude error between the altitude of the UAV and a constant desired altitude, and  $u(t)$  is the throttle input difference from the trimmed equilibrium throttle input.  $p_1(\dot{e}(t))$  and  $p_2(\dot{e}(t))$  are polynomial elements including the aerodynamics generated by the canopy of the UAV and are described as

$$\begin{aligned} p_1(\dot{e}(t)) = & 6.270 \cdot 10^{-4} \cdot \dot{e}^2(t) + 7.271 \cdot 10^{-2} \dot{e}(t), \\ p_2(\dot{e}(t)) = & 1.188 \cdot 10^{-4} \cdot \dot{e}^2(t) - 7.358 \cdot 10^{-3} \dot{e}(t). \end{aligned}$$

We consider two kinds of uncertainties. The first uncertainty is aerodynamics uncertainty, i.e.,  $\Delta_\alpha(t)$ , since it is very difficult to exactly obtain the real aerodynamics of the canopy. The second uncertainty is input uncertainty, i.e.,  $\Delta_\theta(t)$ , since the thrust force generated by a motor is influenced by battery condition, wind conditions, and so on.

Using the sector nonlinearity technique [2], the nonlinear system (76) with the uncertainties is exactly converted into the following four-rule polynomial fuzzy system with uncertainties:

$$\begin{aligned} \dot{\mathbf{x}} = \sum_{i=1}^r h_i(z) \{ (\mathbf{A}_i(\mathbf{x}) + \mathbf{D}_{ai}(\mathbf{x}) \Delta_{ai}(\mathbf{x}) \mathbf{E}_{ai}(\mathbf{x})) \mathbf{x} \\ + (\mathbf{B}(\mathbf{x}) + \mathbf{D}_b(\mathbf{x}) \Delta_b(\mathbf{x}) \mathbf{E}_b(\mathbf{x})) \mathbf{u} \}, \end{aligned}$$

where  $r=4$ ,  $\hat{\mathbf{x}}(\mathbf{x}) = \mathbf{x} = [\dot{e}, e]$ ,  $\mathbf{z} = \dot{e}$  and

$$\mathbf{A}_1(\mathbf{x}) = \begin{bmatrix} p_1(\dot{e}(t)) + p_2(\dot{e}(t)) & 0 \\ 1 & 0 \end{bmatrix},$$

$$\mathbf{A}_2(\mathbf{x}) = \begin{bmatrix} p_1(\dot{e}(t)) - p_2(\dot{e}(t)) & 0 \\ 1 & 0 \end{bmatrix},$$

$$\mathbf{A}_3(\mathbf{x}) = \begin{bmatrix} -p_1(\dot{e}(t)) + p_2(\dot{e}(t)) & 0 \\ 1 & 0 \end{bmatrix},$$

$$\mathbf{A}_4(\mathbf{x}) = \begin{bmatrix} -p_1(\dot{e}(t)) - p_2(\dot{e}(t)) & 0 \\ 1 & 0 \end{bmatrix},$$

$$\mathbf{B}(\mathbf{x}) = \begin{bmatrix} 0.1336 \\ 0 \end{bmatrix},$$

$$\mathbf{D}_{a1}(\mathbf{x}) = \begin{bmatrix} c_2 p_1(\dot{e}(t)) & c_2 p_2(\dot{e}(t)) \\ 0 & 0 \end{bmatrix},$$

$$\mathbf{D}_{a2}(\mathbf{x}) = \begin{bmatrix} c_2 p_1(\dot{e}(t)) & -c_2 p_2(\dot{e}(t)) \\ 0 & 0 \end{bmatrix},$$

$$\mathbf{D}_{a3}(\mathbf{x}) = \begin{bmatrix} -c_2 p_1(\dot{e}(t)) & c_2 p_2(\dot{e}(t)) \\ 0 & 0 \end{bmatrix},$$

$$\mathbf{D}_{a4}(\mathbf{x}) = \begin{bmatrix} -c_2 p_1(\dot{e}(t)) & -c_2 p_2(\dot{e}(t)) \\ 0 & 0 \end{bmatrix},$$

$$\Delta_{a1}(\mathbf{x}) = \Delta_{a2}(\mathbf{x}) = \Delta_{a3}(\mathbf{x}) = \Delta_{a4}(\mathbf{x})$$

$$= \frac{1}{c_2} \begin{bmatrix} \Delta_a(t) & 0 \\ 0 & \Delta_a(t) \end{bmatrix},$$

$$\mathbf{E}_{a1}(\mathbf{x}) = \mathbf{E}_{a2}(\mathbf{x}) = \mathbf{E}_{a3}(\mathbf{x}) = \mathbf{E}_{a4}(\mathbf{x}) = \begin{bmatrix} 1 & 0 \\ 1 & 0 \end{bmatrix},$$

$$\mathbf{D}_b(\mathbf{x}) = \begin{bmatrix} 0.1336c_1 \\ 0 \end{bmatrix},$$

$$\Delta_b(\mathbf{x}) = \Delta_\theta(t)/c_1,$$

$$\mathbf{E}_b(\mathbf{x}) = 1,$$

$$h_1(\mathbf{z}) = \frac{(\cos(0.2\dot{e}) + 1)(\sin(0.2\dot{e}) + 1)}{4},$$

$$h_2(\mathbf{z}) = \frac{(\cos(0.2\dot{e}) + 1)(1 - \sin(0.2\dot{e}))}{4},$$

$$h_3(\mathbf{z}) = \frac{(1 - \cos(0.2\dot{e}))(\sin(0.2\dot{e}) + 1)}{4},$$

$$h_4(\mathbf{z}) = \frac{(1 - \cos(0.2\dot{e}))(1 - \sin(0.2\dot{e}))}{4}.$$

Since  $\|\Delta_{ai}(\mathbf{x})\| \leq 1$  for  $i = 1, \dots, 4$  and  $\|\Delta_b(\mathbf{x})\| \leq 1$  as well as in the previous two examples, we have  $\beta_{ai} = 1$  for  $i = 1, \dots, 4$  and  $\beta_b = 1$ . For any situation satisfying  $c_1 > 0$  and  $c_2 > 0$ , no solutions can be obtained by applying the globally robust control design proposed in [36]. Also, no solutions can be found by applying the globally robust control design of the proposed Theorem 1.

Assume that the operation domain for the UAV system is  $D_o = \{\dot{e} : -1.5 \leq \dot{e} \leq 1.5 \text{ and } e : -10 \leq e \leq 10\}$ , the input constraint is  $-5 \leq u \leq 5$ , and  $c_1 = c_2 = 0.1$ . Moreover, the algorithm for solving Theorem 5 is carried out with the initial setting  $\epsilon_g = 0.005$ ,  $\epsilon_F = 0.005$ ,  $\epsilon_V = 0.005$ ,  $\epsilon_L = 0.005$ ,  $\lambda^{\min} = 0.5$ ,  $\lambda^{\max} = 2$ ,  $\Delta\lambda = 0.5$ ,  $p_i^{\min} = 0.2$ ,  $p_i^{\max} = 1$ ,  $\Delta p_i = 0.4$  for  $i = 1, 2$ . By solving the conditions in Theorem 5, a feasible solution is found as

$$\alpha = -0.0027, \quad \lambda = 5.8979,$$

$$V(\mathbf{x}) = 2.8789x_1^2 + 0.1894x_1x_2 + 0.0683x_2^2$$

$$\mathbf{F}_1(\mathbf{x}) = [F_1^{11}(\mathbf{x}) \ F_1^{12}(\mathbf{x})],$$

$$F_1^{11}(\mathbf{x}) = 0.2919x_1^2 - 0.02140x_1x_2 + 0.0016x_2^2$$

$$+ 0.2850x_1 + 0.0073x_2 + 1.2042,$$

$$F_1^{12}(\mathbf{x}) = -0.0214x_1^2 + 0.0016x_1x_2 - 0.0002x_2^2$$

$$+ 0.0073x_1 + 1.3281 \times 10^{-5}x_2 + 0.18254,$$

$$\mathbf{F}_2(\mathbf{x}) = [F_2^{11}(\mathbf{x}) \ F_2^{12}(\mathbf{x})],$$

$$F_2^{11}(\mathbf{x}) = 0.2677x_1^2 - 0.0231x_1x_2 + 0.0012x_2^2$$

$$+ 0.3501x_1 + 0.0125x_2 + 1.3162,$$

$$F_2^{12}(\mathbf{x}) = -0.0231x_1^2 + 0.0012x_1x_2 - 0.0002x_2^2$$

$$+ 0.0125x_1 + 6.3453 \times 10^{-5}x_2 + 0.1877,$$

$$\mathbf{F}_3(\mathbf{x}) = [F_3^{11}(\mathbf{x}) \ F_3^{12}(\mathbf{x})],$$

$$F_3^{11}(\mathbf{x}) = 0.266x_1^2 - 0.02187x_1x_2 + 0.0012x_2^2$$

$$- 0.3234x_1 - 0.0116x_2 + 1.3234,$$

$$F_3^{12}(\mathbf{x}) = -0.0219x_1^2 + 0.0012x_1x_2 - 0.0002x_2^2$$

$$- 0.0116x_1 - 0.0002x_2 + 0.1868,$$

$$\mathbf{F}_4(\mathbf{x}) = [F_4^{11}(\mathbf{x}) \ F_4^{12}(\mathbf{x})],$$

$$F_4^{11}(\mathbf{x}) = 0.2813x_1^2 - 0.0212x_1x_2 + 0.0016x_2^2$$

$$- 0.2650x_1 - 0.0078x_2 + 1.2181,$$

$$F_4^{12}(\mathbf{x}) = -0.0212x_1^2 + 0.0016x_1x_2 - 0.0002x_2^2$$

$$- 0.0078x_1 - 0.0002x_2 + 0.18,$$

etc. Moreover, by solving the optimization problem of (60) and (61), the outmost Lyapunov function level set contained in the operation domain is obtained as  $\Omega_{V,6.1822} = \{\mathbf{x} : V(\mathbf{x}) \leq 6.1822\}$ , and the SOS multipliers are obtained as  $\phi_1(\mathbf{x}) = 0.3640$  and  $\phi_2(\mathbf{x}) = 15.676$ . Fig. 8 shows the outmost Lyapunov function level set (the contractively invariant set) and the controlled results for six cases of initial states. From Fig. 8, the control system is asymptotically stable and the  $\Omega_{V,6.1822} = \{\mathbf{x} : V(\mathbf{x}) \leq 6.1822\}$  is a contractively invariant set. Fig. 9 shows the control inputs for the six cases. It can be seen from Fig. 9 that all control inputs satisfy the input constraint  $-5 \leq u(t) \leq 5$ . Fig. 10 shows the time response for Case 1. Fig. 11 shows the control input for Case 1.

## VIII. CONCLUSIONS

This paper has presented a new SOS design framework for robust control of polynomial fuzzy systems with uncertainties. Two kinds of robust stabilization conditions, i.e., global SOS robust stabilization conditions and semi-global SOS robust stabilization conditions, are derived in terms of SOS. The new design framework has given key ideas to avoid conservative issues. The first key idea is that we directly solve non-convex SOS design conditions without applying the typical transformation. The second key idea is that we bring a so-called copositivity concept. These ideas provide some advantages in addition to relaxations. To solve our SOS robust stabilization conditions efficiently, we have introduced a gradient algorithm formulated as a minimizing optimization problem of the upper bound of the time derivative of polynomial Lyapunov functions. Three design examples have been provided to illustrate the validity and applicability of the proposed design framework. The examples have demonstrated advantages of the new SOS

design framework for the existing LMI approaches and the existing convex SOS approach. Our next subjects are to apply the advanced SOS robust stabilization conditions to more complex systems, e.g., [44], [45], [46].

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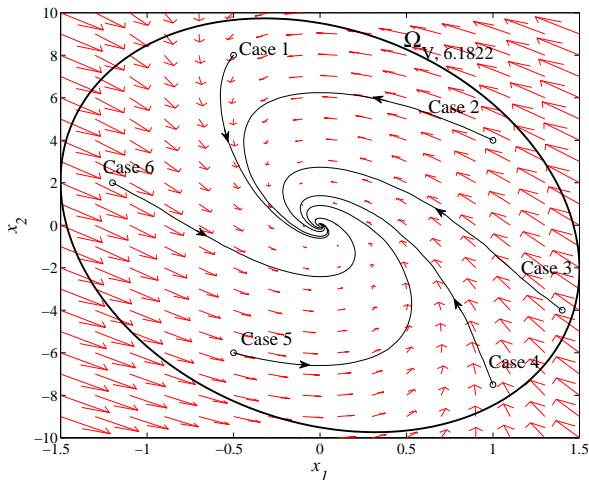


Fig. 8. The controlled results for six cases of initial states.

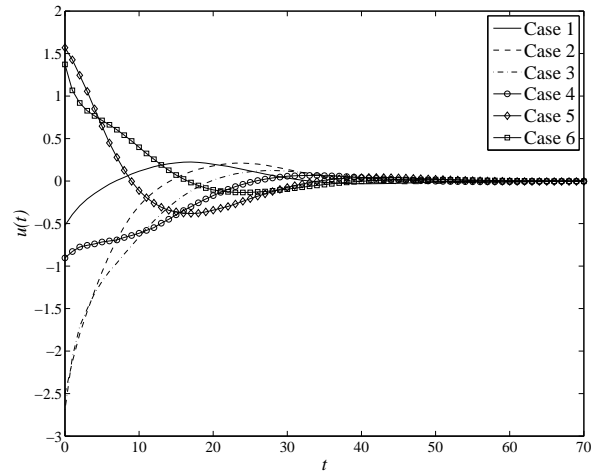


Fig. 9. The control inputs for the six cases.

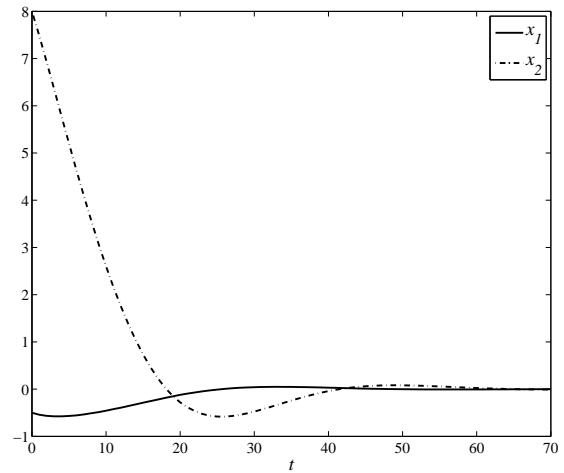


Fig. 10. Time response for Case 1.

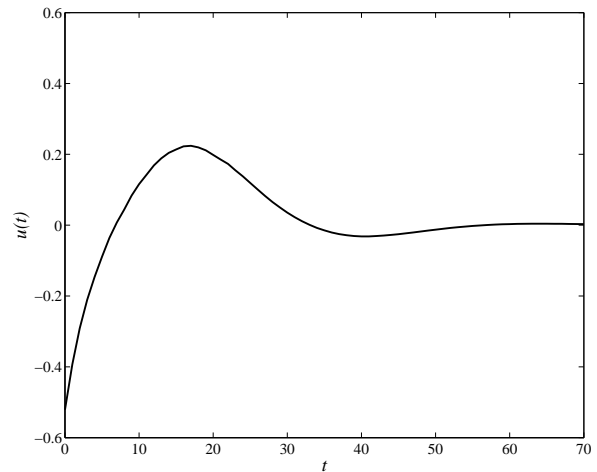
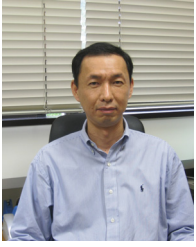


Fig. 11. The control input for Case 1.

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